Registration: 8356 Date: 07-06-05		Manufacture: Serial No:	IAI 868	Model I Entered	to: 1124 By: JKG	
otal Time:	The second secon	al Landings:		Work Or	der No.	Item No
				07-0	25	
Discrepancy: Com	LEY WIT	H COCKI	PIT ANI	D CABIA	TIRES	OTTLE
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In accordance with 14	CED 5 42 0 /5	42 44 (nan/a) (and that the	energy and the		
herein has been done						
applicable) recommen	nded methods.	and procedures	contained within	the manufactu	ire's current	maintenance
manual. The subseque which it was complete						
maintenance/inspection	ons performed	in accordance wi	th 14 CFR § 13	5.411(a)(2), ref	erence the c	
	proved program			etum to service	Agency for a fight of Agent State of the party of the second seco	
certificate holders app		C A Phy bro	TOMMINA ATOM		1.1.04.0	3
certificate holders app Technician St			cate Number		Date	2
Techandan Si	mature	A&P 393	75/8	\$	7-06-9	25-
	mature	A&P 393	cate Number	ø,		25 -

Westwind 1124

Pre and Post Flight inspection N8356

Nose torque link disconnected Lav serviced	
LPB Pitot and static covers installed	
Gear and tires for condition	
Engine oil levels (1 qt low)	
Signature Jang P. Bauto	Date 07-27-2005
그리즘 그는 그런데 회사를 하는데 하는 사람들은 그는 그는 그는 그는 그를 가는 것이 되었다.	
Pre Flight	
(1B)	
Windshields and cabin windows for condition	
Wings, de-ice boots and stall strip for condition	
Tip tank fuel valve closed (up)	
Allerons, flaps and speed brakes for condition	
HP repture membrane in place	
Hyd reservoir level	
Gear extension cylinder (1700-2000 psi)	
Thrust reverser cylinder (710-940 psi)	
Engines for leaks	
Engir e inlets and exhaust for FOD	
Exterior lights	
Interior systems checked	
Oxygen full (1800-2000 psi) SERVICEI)	
Brake wear	
Tire inflation (nose 55psi main 155 psi) JEK	VICEI
Sump fuel	-,
Signature Sams. Darres	Date 07-77-2005



el Time: 8			રેલ્ક	Entered	By: SM	
	196.6 Total Lan	dings:	6342	Work Ord	er No.	Item No
Language of the control of the contr				07-0	05	
crepancy: La	t MAIN TH	RE Wo	RN			
			·			
rrective Action:	LH MAIN WA	HRC'ASS	1 REMOU	es and	Date.	Tech I
	O WOTH BUIL					SM
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Removed	Reinstalled	Date	Function CK	Date	Leak CK	DI
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TIPE	249K 83-3	41560	920	249 K 83-3	100	601065
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147"

9000 Randolph St. Houston, Texas 77061 713-644-1128

Description WEST WIND MAIN No SOCIETY PINTE 244K83-3 SINDRE-SO	N 03-890
	101065 130
Date Remarks WH	EEL WAS ASSY. US
AIRCRAFT BRAKING SY	TEME COMM. TIRE

	oving National Aviation	2.			· · · · · · · · · · · · · · · · · · ·			3. Fo	rm Tracking Number:
	hority/Country: /United States	AUTI		D RELF 130–3, AIRWOR			FICATE		406495
AERO-L 405 NEI SOUTH	eation Name and Address: IFT EQUIPMENT COURTS BRASKA HOUSTON, TX 775 PAIR STATION XV1	OMPANY 587	raa rum o	130-3, ARWON	TINVESS A	1		5. W Num	ork Order/Contract/Invoice ber: 406495
6. Item:	7. Description:	8. Part	Number:	9. Eligibi	lity: *	10. Quantity	: 11. Serial/Batch Nu	mber:	12. Status/Work:
1	WESTWIND BOL	TS PN: G	Y187-36	N/A		8	N/A		REPAIRED
2	WESTWIND NUT		GYN187	N/A		8	N/A		REPAIRED
*NOTI	BLOCKS 6, 7, 8, 10, ANI UNDER THE WORK OR ACCORDANCE WITH FI DATA/CMM. INSPECTI MAGNETIC PARTIC	D 11 AS APPLICAB DER AND SYSTEM EDERAL AVIATION ED IN ACCORDANC CLE INSPECTIO	LE. A COMPLET TRACKING REF ADMINISTRATIC E WITH ASTM14 N — THERE W s) described hereon af	TE DESCRIPTION ERENCE NUMBE ON, PRODUCTION 144 FOR MAGNET / ERE NO MAGNET OF THE BOTTO	OF WORK P R INDICATEI I APPROVAL TIC PARTICL INETIC DIS	ERFORMED IS D IN BLOCKS : L HOLDERS AP E INSPECTION SCONTINUI	ON FILE AT THE ABOV AND 5. THE DESCRIBE ID/OR THE MANUFACTU . (STARFLITE FIES FOUND IN THE ISSURING COMPILENCE WITH	E REFE ED WOR IRERS MAN E ABO'	DESCRIPTION LISTED IN ERENCED ORGANIZATION RK WAS PERFORMED IN APPROVED TECHNICAL AGEMENT GROUP) WE PARTS Discribe Airworthiness Directives ulation specified in Block 13
	Approved design data and Non-approved design data	l are in a condition fo	r safe operation.		and des Federa	scribed in Block	nerwise specified in Block 1 13 was accomplished in account 43 and in respect to that	cordanc	e with Title 14, Code of
15. Autho	rized Signature:		16. Approval/A	uthorization No.:	20. Authori	zen Signature:		1	Approval/Certificate No.: 11R626K
17. Name	(Typed or Printed):		18. Date (m/d/y)): /	22. Name (1 JACK EL	Typed or Printed):		Date (m/d/y): 07-2005
		-		User/Installer l	L	ities		!	
Where the Block 1, it Block 1.	e user/installer performs w is essential that the user/ii	ork in accordance winstaller ensures that h	cument alone does th the national reg his/her airworthine	not automatically of ulations of an airwess authority accept	constitute auth orthiness auth is parts/compo	ority to install t ority different t nents/assemblie	he part/component/assemb nan the airworthiness autho s from the airworthiness au ntain an installation certific	ority of t ithority	the country specified in of the country specified in sued in accordance with the

Registration: N83 Date: 121		anufacture: erial No:	1A1 368	Model N Entered		24 HG.	
Total Time: 874	the same of the sa	andings:	1342	Work Or	der No.	Ite	m No.
8390.5/6536	8688.6/610	99		67-0	5		
Discrepancy: F	VEL CEAK	x - R1	SHT U	1116			
Corrective Action:	defueled & o	pered fue	1 panels (?	-2089)	DATE	TECH	TIME
Corrective Action: (sealant &	removed	rive/s(7	2183	7-20	82	6.0 V
					7-21	87	7.0
				· · · · · · · · · · · · · · · · · · ·	7-22	80	70
MI					17-11	1900	1.0
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& Repla	29513-0 29513-0 ced 4 scre	ws my	24694	\$ 99			
	*						
Removed Date	Reinstalled	Date	Function CK	Date	Leak CK		Date
Pasition	P/N OFF	S/N	OFF	P/N ON		S/N O	N
1 33.1131		0111		THEOR		3/14 0	
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In accordance with 14 C herein has been done sa	FR § 43.9 (§ 43.1 atisfactorily using	<pre>11-Insp's) ce the aircrafts a</pre>	ertify that the wirframe, endin	vork described, p e. or appliance	performed manufact	d, and cor ure's (as	npleted
applicable) recommende	ed methods, and p	procedures co	ntained within	the manufactur	e's currer	t mainter	nance
manual The subsequer which it was completed.	constitutes a retu	irn to service of	only for the wo	ork performed ar	nd comple	eted herei	
maintenance inspection	s performed in ac	cordance with	14 CFR § 13	5.411(a)(2) refe	rence the	current	
certificate holders appro Technician Signa	ature		ine required re t <mark>e Numbe</mark> r	num to service :	statement Da		
8000 7	Lynch A8	2668	256	-	- 22	-05	
Inspector Signa	fore A		te Number		Da		
4 minh	JC-76	2) Until	9161	1	11	1	
Mull N. O.	MAG / 119 4 9	7 73491	019160		1001	05	

Section 10 00

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Maintenance Work Order

Registration: 18	1356	Manufacture: 1A1 Serial No: 368			4	
Total Time: 8796	le Tota	Landings: 634	Work C	rder No.	Item	No.
8390.5/5970	8688.4	/6199	0-	7-05		
Discrepancy: 20	EFT WII	VA ROOT -	RIVETS D.	MAGE	0.?	
Corrective Action:	Repaire	d by Fligh	folia	7-20		TIME 2.5
		7-20	or			
FCIGHT FAACUS F Removed Date	- VEHICLES -8VR2340 Reinstalled	Wolf (CG) Date Fund	Cotion CK Date	Leak CK		Date
Position	P/N OFF	S/N OFF	P/N ON		S/N ON	1
herein has been done applicable? recommer manual. The subsequi which it was complete maintenance/inspection	satisfactorily usinded methods, and ent signature, will ed, constitutes a rooms performed in proved program a	3.11-Insp's) I certify thing the aircrafts airframed procedures contained high certificate number appeturn to service only for accordance with 14 CF and/or GMM for the reconcertificate Number 14 CF	e, engine, or appliance within the manufact operate to the work or the work performed of \$135.411(a)(2) required return to service.	e manufacture's curren performed vand comple ference the	ure's (as t maintena with the da ted herein current	ance ite on
Cauil N. L	ent.	<u>Certificate Num</u> A&9 45-998/9/6/	ber	Da 1/20/0	<u>te</u>	mass to should be

Section 10 00

		rial No:	1H1 368	Model I Entered	1 / 60	OK OK	
otal Time: 8196	. 6 Total La	andings:	6342	Work O	rder No.	lte	m No.
8390,5/59	70 848	8.6/6199		07-	04		
CABIN SE	AKE OFF AT + TI	+ LAN	DING PR POSITION	ACARD: (S - M)	SSINC	<u></u>	
orrective Action:	Installed pla	cards on	tables		DATE	TECH	TIME
					720	10	.5
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Removed Date	Reinställed	Date	Function CK	Date	Leak CK		Date
Removed Date Position	Reinstalled P/N OFF		Function CK OFF	Date P/N ON	Leak CK	S/N O	
	P/N OFF P/N	1-Insp's) I ce the aircrafts a procedures co ertificate num rn to service o cordance with for GMM, for t	off entify that the woirframe, engine ntained within the propriete only for the wor 14 CFR § 135.	P/N ON Drk described, a, or appliance the manufactu a to the work p k performed a 411(a)(2), refe	performed manufacture's curren performed vand comple erence the statement	S/N O	npleted nance late on
Position accordance with 14 Cerein has been done sopplicable) recommend anual. The subsequer hich it was completed aintenance inspection ertificate holders appropried appropriate the subsection of the subsecti	P/N OFF P/N	1-Insp's) I ce the aircrafts a procedures co ertificate num in to service o cordance with for GMM, for the Certificate	off ertify that the woirframe, engine ntained within taber appropriate only for the wor 14 CFR § 135. the required retile Number	P/N ON Drk described, a, or appliance the manufactu a to the work p k performed a 411(a)(2), refe	performed manufacture's current performed vand completerence the	S/N O	npleted nance late on
Position accordance with 14 Cerein has been done sopicable) recommend annual. The subsequer hich it was completed aintenance/inspection ortificate holders appropries.	P/N OFF OFR § 43.9 (§ 43.1 atisfactorily using the distribution of the constitutes a return of the constitutes are the constitute of the constitution of the	1-Insp's) I ce the aircrafts a procedures co ertificate num rn to service o cordance with lor GMM, for I Certificat	off ertify that the woirframe, engine ntained within taber appropriate only for the wor 14 CFR § 135. the required retile Number	P/N ON Drk described, a, or appliance the manufactu a to the work p k performed a 411(a)(2), refe	performed manufacture's curren performed vand comple erence the statement	s/N O	npleted nance late on

Section 10 00

Registration: N8356 Date: 7/20/05	Manufacture: Serial No:	1A1 3108	Model N Entered	The first	7K.	
Total Time: 8796.4	Total Landings:	6342	Work Or	der No.	lter	n No.
8390,5/5920	3688.6/6199		07-0	05-		
Discrepancy: LEFT	HORIZONTI	AL DE-1	CE BU	197 (d	EAKIN	G.
Corrective Action: Sealed (seal b	LH deice	boot leak	through the State of the State	7-20	TECH	TIME 10
				701	337	
Removed Date Rem	nstalled Date	Function CK	Date	Leak CK		Date
Position P/N	OFF SIN	OFF	P/N ON		S/N O	N
In accordance with 14 CFR § 43. herein has been done satisfactor applicable) recommended metho manual. The subsequent signatus which it was completed, constitut maintenance inspections perform certificate holders approved progressing approved progressing Signature. Language Constitution of the Constituti	ily using the aircrafts a cs, and procedures co re, with certificate num es a return to service ched in accordance with ram and/or GMM, for Certificate A&P 2660 Certificate	irframe, engine, ntained within the ber appropriate only for the work 14 CFR § 135.4 the required return to Number	or appliance be manufacture to the work per performed are [11(a)(2), refe	manufacture's currence formed with the comple rence the statement Date 7-20	ure's (as t mainten with the d ted herein current	ance ate on n. For

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Registration: N 835G Date: 7/20/05	Manu: Serial	facture: No:	1A1 368	Model N Entered		4	
rotal Time: 879/2 Le	Total Land	lings:	6342	Work Or	der No.	Ite	m No.
8390.5/5910	8688.6/610	39			-05		
Discrepancy: NOSE	GEAR	Tou	CIMIT	MARKIN	14/1	RACIAR	0.
Corrective Action:	PCARDS	000	ERED.		DATE	TECH	TIME
Removed old	placard	ł i	ustalled N	ew one	7-20	32	.5
			304 (200 - 2				
Removed Date R	einstalled	Date	Function CK	Date	Leak CK		Date
Position P/N	NOFF	S/N	OFF	P/N ON		S/N O	N
Metallia interpreta in the control of the control o	CC SATURATION AND CONTRACTION OF THE CONTRACTION OF	CONTRACTOR STATE AND STATE AND STATE AND		of these surprises property and the second		Control of the Contro	
n accordance with 14 CFR § 43 erein has been done satisfacto pplicable) recommended methodoxial. The subsequent signat which it was completed, constitutional transportance inspections performatificate holders approved pro-	orily using the nods, and procure, with certing the time of the term of term of term of the term of th	aircrafts a cedures co- ficate num to service of dance with	irframe, engine, ntained within the ber appropriate only for the work 14 CFR § 135.4	or appliance ne manufactur to the work p performed a \$11(a)(2), refe	manufactore's current erformed of and completerence the	ure's (as it mainter with the d sted herei current	nance late on
Technician Signature	ogram andror		e Number	in to service	<u>Da</u>		
San Lynn	/ A&P	266	9256		7-20	-05	-
Inspector Signature			e Number		Da		
Dayil N. Lent	(82	X599,	81960		7/201	105-	
Section 10 00	-4						Page 38

Registration:	83\$G	Manufacture:	IAI	Model No:		24
Date:	7-25-05	Serial No:	368	Entered B	STREET, SQUARE,	
Total Time:	8796.6 TO	tal Landings:	6342	Work Orde	r No.	Item No.
8390,5	5970	688,6/6199		07-05		
Discrepancy:	LH BRH	Hor. stab	de-ice bo	ots domage	1	
-						
Corrective Actio	n: Repaired K.t.	de danage	lareas wi		7-25	ECH TIME
				7	7-26 7-27	82 3,5 90 1.0 10 1.0
					7-11/2	PB 1.5
	× =					
Removed	Date Reinstal	led Date	Function CK	Date	Leak CK	Dale
Position	P/N OF	F S/N	OFF	P/N ON		S/N ON
	40					
herein has been of applicable) recommendal. The sub-	th 14 CFR § 43.9 (§ done satisfactorily to mended methods, sequent signature,	ising the aircrafts a and procedures co with certificate num	airframe, engine, ontained within the ober appropriate	or appliance mane manufacture's to the work perfe	anufacture current mormed with	is (as naintenance n the date on
maintenance insporting	pieled, constitutes i sections performed s approved program n Signature	in accordance with and/or GMM_for Certifica	14 CFR § 135.4 the required retu te Number	111(a)(2), referen	nce the cu	rrent
-Sean Inspector	Signature	A&P2 868	256 ite Number	7	-26-0 Date	55
Daniel	V. Lende.	A&P 45981	940 (7	126/0	5



Registration N 8356 Date: 07-21-2005	Manufacture: TAI Serial No: 348	Model No: Entered By:	A C
	tal Landings:	Work Order No	o. Item No.
Discrepancy PRE 190	ST FLIGHT		
Correct Settlers COMPLE FITSHT CHECK OXYGEN AND	Y WITHPRE! LIST. SERVICE ALL FOUR TI	PBST DAT	
Remilies Cate Rumer's	ec Date Function CK	Oaie Lea	k CK Date
Position P/N OF	F S/N OFF	P/N ON	S/N ON
s activos i troja	ising the aircrafts airframe, engine, and procedures contained within th with tertificate number appropriate	or appliance manufice manufacture's cuito the work perform performed and condition (2) reference in to service statem	rent maintenance ed with the date or noieted herein. For the current nent. Date 7 - 2 005
5. W. SHV. L	A&P		<u>Date</u>

Registration N 8 3 5 6 Date: 07-29-2005	Manufacture: IAT Serial No: 308	And the same of th	o: 1124 A By: LPB	
	otal Landings:	Work Ord		n No.
Discrepancy DELIVER	TO DUNCAN	FOR R	V.5.M	
AVZATION FO	D TO RATEA	NTIC	7-29 LAB 7-29 Sm	1.0 (.5
Remiyes Date Pens	ralled Date Function CK	Date	Leak CK	Dale
Pas tien P/N O	FF S/N OFF	P/N ON	S/N OI	N
In eccarcian a win 14 CFR § 43.9 here is has been signed satisfectorily applicable in commenced method manual. The subsequent signature which is wis manufeled constitute maintenance of sections can be applied to a section of sections. The short of Signature. Section of Signature.	ising the aircrafts airframe, enging and procedures contained within with certificate number appropria	ne, or appliance read the manufacture at the manufacture at the work performed an install (a)(2) refereturn to service s	manufacture's (as e's current mainten erformed with the d ed completed herein rence the current	ance ate or

Registration: N83S6 Date: 7-28-05	Manufacture: TA		110
Anna (1974)	tal Landings:		rder No. Item No.
Discrepancy: Left &			
Corrective Action: found Removed ignite S/H 4/45/67 and 3070378-2 s ck Hamal	bad 1g wite on box pp sd1, wholl igual A 90065805	1 box 3 070378-2 ter box ph , 6.Ps	DATE TECH TIME
Removed Date Reinsta	led Date Fur	action CK Date	Leak CK Date
Position P/N OF	F S/N OFF	P/N ON	S/N ON
In accordance with 14 CFR § 43.9 (§ herein has been done satisfactorily (§ 43.11-Insp's) I certify t	hat the work described,	performed, and completed
applicable recommended methods, manual. The subsequent signature, which it was completed, constitutes maintenance inspections performed certificate holders approved program Technician Signature	and procedures containe with certificate number a a return to service only fi in accordance with 14 C	ed within the manufacture ppropriate to the work por the work performed a FR § 135.411(a)(2), refugured return to service amber	ure's current maintenance performed with the date on and completed herein. For ference the current
Inspector Signature	Certificate Nu	mber	Date
	A&P		

4. Approv	pional aviation authority / cour ransport Canada ed organization name and address ywell Aerospatiale Inc.	AUTHORIZ	TCCA 24-00		5. WO VADA H5-	756142 NEYWELL AEROSPACE SERVI ork order / Contract / Invoice 6528558 D: 373757000
B. Item	7. Description	8. Part No.	9. Eligibility *	10. Qty	11.Serial/Batch No.	12. Status/Work
1	EXCITER, IGNITION	3070378-2	VARIOUS	1	90065805	OVERHAUL
		(LAST PAGE OF DOC.)				PAGE 1
	CMM 74-10-3	8,REV.2.FEB 28 1993		ork are on the file at this A		r retum to service. Pertinent details Organization Acceptance # EASA.145.7010
20cadi	ion that the items identified above	were manufactured in conformity to :		19.		Acceptance # EASA.145.7010
14. OEIII		e in condition for safe operation.		CAR 571		ock 13. block13, the work identified in block 12
		_		20. Authorized signature	AVX	1 21. Certificate / Approval ref No.
15. Authoriz	ed signature	76-Cartificate / Approval n	1-72	9	man (33)	1-7

3. Statements 14 and 19 do not constitue installation certification. In all cases, the aircraft technical record must contain an installation certification, issued in accordance with the national regulations

authority accepts products or maintenance from the airworthiness authority specified in block 1.

of the state of registry, before the aircraft may be flown.

3. Form tracking No.

Honeywell

Honeywell Aérospatiale Inc.

RAPPORT DE CONDITION / CONDITION REPORT

	CLIENT / CUSTOMER:	HONEYWELL AEROSPACE	# BON DE TRAVAIL / WORK ORDER #:	T373757
T	DESCRIPTION:	EXCITER, IGNITION	DATE:	16 MAY 2005
	# DE L'UNITÉ REÇU /		# SÉRIE REÇU /	jour/day mois/month année/year
	PART # RECEIVED:	3070378-2	SERIAL # RECEIVED	90065805
2	RAISON DU RETOUR / REASO	ON FOR RETURN:		
,	OVERHAUL		127	
3	INSPECTION VISUELLE EXTE	RNE / EXTERNAL VISUAL INSPECTION:	n sing	
	NORMAL			
4	RÉSULTATS DU TEST INITIAL INCOMMING TEST RESULTS:	. <u>/</u> N/A	DANS LES LIMITES / WITHIN LIMITS	HORS DES LIMITES / OUT OF LIMITS
	COMMENTAIRES - OBSERVA	TIONS / COMMENTS - OBSERVATIONS:		·
T'1	NO OUTPUT			
granie de la Granie		1.6		
5		ÉMONTAGE / DISMANTLING CONDITION		
	SPARK GAP AND STOR BURN MARKS CREATES	AGE CAPACITOR BREAKDOWN S SHORT (WIRES REPLACED). SI	MOOTHING CAPACITOR N	IOT GOOD.
	RIPTION DE LA PIÈCE / DESCRIPTION	# DE PIÈCE / PART #	COMMENTAIRES - OBSERVATION	
	THING CAPACITOR	SPR135D117X9075K6 (10-392899)	NOT GOOD	
			da.	
<u></u>				
			- 17	· .
ho t	TO THE PROPERTY OF THE PROPERT	4 54 10 20		
6	PUBLICATION TECHNIQUE TECHNICAL PUBLICATION	/ 74-10-38 REV 2, 28-FEB-1993		
7		E / SERVICE BULLETINS		
SB		SB FD	Révisé / Overhaul	Réparé / Repaired
SB		SB	, e	
SB		SB	Modifié / Modified	Inspecté / Inspected
SB		SB		
8	Numéro de l'unité : Part Number :		Numéro de Série : Serial Number :	
	Bloc #8 à remplir seul	ement si diffèrent du bloc #1 / [Block #8 to be filled only if di	ifferent from block #1
õ41	B Ilel	me	I Tramed	au (4711)
	,	TECHNICIAN	RÉVISÉ PAR / REVIEV (nom & # employé) / (name & e	
ı	(nom & # employé) / (r	name & employee #}	(nom & # employe) / (name & e	on project at

Printed: May 12, 2005

-Honeywell

Honeywell Aérospatiale Inc.

ODRN No. 2246-5

"OPAID" OVERHAUL PROCESS AND INSPECTION DOCUMENT

PART NO.

NSN:

NO. DE PIÈCE:

10-392000-1 (3070378-1)

10-392000-2 (3070378-2)

COMMERCIAL

DESCRIPTION:

Ignition Exciter

REF. SPEC.

DOC DE RÉF.

ATA 74-10-38 (Rev 2, 28-Feb-93)

WORK ORDER

BON DE TRAVAIL:

SERIAL NO. / NO. DE SÉRIE:

MFR / FAB:

Unison Industries

OPAID REVIEWED / APPROVED BY: OPAID RÉVISÉ / APPROUVÉ PAR:

T. Carfagnini

DATE:

09-Dec-04

OF REF. SPEC. PAGE " WORK CONTINUED FROM DŁU DOC. DE RÉF. TRAVAIL COMPLÉTÉ À PARTIR DE LA

REVISÉ / REPARÉ / MIS À L'ESSAI PAR:

FINAL ACCEPTANCE BY: ACCEPTATION FINALE PAR:

DATE:

- PRELIMINARY DIAGNOSIS: Carried out as per ref. spec. 1.0
- **DISASSEMBLY AND CLEANING:** 2.0

Carried out as per ref. spec.

DETAILED EXAMINATION, REPAIR AND REASSEMBLY: 3.0

Carried out as per ref. spec.

Ensure the required in-process inspection checks where indicated by a 3.1

are carried out.

4.0 <u>FINAL TESTING</u>: (Ref. ATA 74-10-38, Rev 2)

PARA		TEST		REQUIRED	UNITS	ACTUAL	
3.A	Diode Check	Resistance		0.2 min	megohms	2Ms	
		Resistance between connector Pin 'A'	n exciter input and exciter housing	0.05 max.	ohms	0.0	12.
3.B	Continuity Check	Resistance between exciter input connector Pin 'B' and Pin 'C'		0.05 max.	ohms	0.0	2
		Input Voltage	Input Current	Actual Current	Spark Rate (spark/sec.)	Spark Rate (spark/sec.)	
	Spark Rate and	(V dc)	(Amps)	(Amps)	Left, Right	Left	Right
		Spark Rate and 10 2.0	2.0 max.	1.6	1.0 to 6.0	1.8	1.8
3.C.5	Input Current	24	2.0 max.	1.7 -	1.0 to 6.0	4.3	4.3
Í		30	2.0 max.	1.7 -	1,0 to 6.0	4.6	4.6
		40	2.0 max.	1.8 -	1.0 to 6.0	5,5	5,5
1 D	Input Smoothing	Input Voltage (V de)			Waveform		
3.D	Capacitor		30	Fig. 103 (B)	Ac	ccept 🗖	

PARA	TEST		REQUIRED	UNITS	ACTUAL
		Adjustable Ball Gap	24	kV	
		Applied Input Voltage	10	V dc	
		The ball gap shall not fire consistently	17	Accep	ot 🗹
		Adjustable Ball Gap	18	kV	Section of the second
		The ball gap shall fire consistently for 10	seconds min.	Ассер	ot 🖫
3.E	Output VoltageTest	Repeat test for other output	24	kV	
		Applied Input Voltage	10	V dc	Section 1
		The ball gap shall not fire	•	Accep	ot 🔽
		Adjustable Ball Gap	18	kV	1 1
		The ball gap shall fire consistently for 10	seconds min.	Accep	ot 19

0 FINAL ACCEPTANCE INSPECTION.

- a) All 'OPAID' items completed including mandatory inspections, where applicable.
- b) Check for completion and shipping readiness. Ensure that shipping caps, lockwires and seals are present where required.
- c) Ensure all documents have been completed.

GARRETT AVIATION SERVICES PART IDENT LABEL PART IDENT LABEL APP CERT REMOVED PART DETAIL QUANTITY U.O.M PART NUMBER EA 5152882 1 SERIAL NUMBER PART CODE REMOVAL TYPE SERIAL NUMBER SCHED/UNSCHED 74 0 26 90065805 PART NUMBER: 3070378-2 DESCRIPTION DESCPN: IGNITION UNIT STATION POSITION A/C DATE SHELF LIFE EXPIRY CONDITION OWNER 0/00/0000 P0001 В BIN NO. ORIGINAL BATCH LOC REASON FOR REMOVAL CONDITION 0107 5152882 IAH SERV. FITTED TO VENDOR ORDER NUMBER U/S 1114799 TECH LOG PAGE INSP STAMP DATE STAMP/SIGNATURE DATE 6/24/2005 RECEIVED DATE:

VS/QA 025 ISSUE 2



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STARFLITE AVIATION General Maintenance Manual

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4. Organization Name and Address: AERO-LIFT EQUIPMENT COMPANY 406 NEBRASKA SOUTH HOUSTON, TX 77587 FAA REPAIRS STATION, XV1R626K 713-947-2963 6. Item: 7. Description: 8. Part Number: 9. Eligibility: * 10. Quantity: 11. Serial/Batch Number: 12. Status/Work: 1 BOLTS GYN187 8 N/A REPAIRED 13. Remarks: A GENERAL DESCRIPTION OF THE WORK PERFORMED IS ATTACHED AS THE TEARDOWN REPORT; UNDER THE PART DESCRIPTION LISTED IN BLOCKS 6, 7, 8, 10, AND 11 AS APPLICABLE. A COMPLETE DESCRIPTION OF WORK PERFORMED IS ON FILE AT THE ABOVE REFERENCED ORGANIZATION ADMINISTRATION, PRODUCTION APPROVAL HOLDERS AND/OR THE MANUFACTURERS APPROVED TECHNICAL CORDANCE WITH FEDERAL AVIATION ADMINISTRATION, PRODUCTION APPROVAL HOLDERS AND/OR THE MANUFACTURERS APPROVED TECHNICAL DATA/CMM. REPAIRED IN ACCORDANCE WITH ABSC CMM 324-037, AP-446. DATA/CMM. REPAIRED IN ACCORDANCE WITH ABSC CMM 324-037, AP-446. DATA/CMM. REPAIRED IN ACCORDANCE WITH ABSC CMM 324-037, AP-446. DATA/CMM. REPAIRED IN ACCORDANCE WITH ABSC CMM 324-037, AP-446. SCTARF-LITE MANUFACTURERS APPROVED TECHNICAL CONTROL WITH A SECONDAY C	FAA	/United States	AUTE	HORIZE FAA Form 81	D KELE 30–3. AIRWOR	LASE (THINESS A	PPROVAL TAC	ICAIL			
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17. Name (Typed or Printed): JACK ELLIOTT 17. Date (Intro). JACK ELLIOTT 17. Date (Intro). JACK ELLIOTT 18. Date (Intro). JACK ELLIOTT 19. Date (Intro).	15. Autho	rized Signature:		16. Approval/Au	thorization No.:	20. Authora	zed Signature:		1		
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Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.	Where the Block 1, it Block 1.	user/installer performs was is essential that the user/i	ork in accordance wi nstaller ensures that l	ith the national regunds. his/her airworthines tion certification. Ir	ılations of an airwo ss authority accept	orthiness auth s parts/compo	ority different tha nents/assemblies f	n the airworthiness author from the airworthiness au	ority of (thority	of the country specified in	

Registration: N 8356	Manufacture: J		Model No:)
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harrin has been done estigiscin	rily using the aircrafts a	airframe, endin	e, or appliance ma	iliniacini e i	5 f 6 2
applicable) recommended method manual. The subsequent signature	ods, and procedures co	ntaineu witnin nber appropria	te to the work perfe	ormed with	the date or
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Inspector Signature		te Number		Date	
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Hauil Atom	4. A&P 4599	X1460	6/2	2/03	

STARFLITE AVIATION General Maintenance Manual

	256		ufacture					11221		
te: 6-24-		Seria	il No:	368		En	tered By	SMG		
te Time:		Total Lan	dings:			Wo	k Order	No.	Item N	ło.
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tegistration: NG		ufacture: IAI	Model No:	1124
late: 5-6.	serie	1 No: 368	Entered By:	Sn
otal Time: 87	So. & Total Lan	dings: 63/4	Work Order N	o. Item No.
			05-05	
discrepancy:	# 1 Engine	FUEL BY FILT	TER BYPASS	
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iz accorcance with	4 CFR § 43.9 (§ 43.1	I-Insp's) I certify that the	work described, perfe	riniec, and complet
applicable) recomm	ended methods, and p	he aircrafts airframe, engi rocedures contained with	n the manufacture's o	current maintenance
manual The subser	ment signature, with co	ertificate number approprie on to service only for the w	ate to the work perior	Wed with the case c
maintenance/inspec	tions performed in acc	cordance with 14 CFR § 18	35.41 1(a)(2), referen	Se the Current
certificate holders a		or GMM, for the required Certificate Number	return to service state	Date
Audit de	Lent. ABF	1 = 2 =	5/4	105
Inspector S	ionatute	Certificate Number	RII	Date
In the	acts A&I	2618848	QC.1) 5-6	-05
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1. Approving Authority/C	Aviation Councy: /United States	I .	THORIZED RELE orm 8130-3, AIRWOR				3. Form Tracking No B05005475	/
4. Organizatio	1	loneywell International, 944 E. Sky Harbor Circl Phoenix, AZ 85034		PRODUCTION APPR PC413	OVAL		5. Work Order/Cont HI3676-003 WOA187163 001	ract/Invoice Number: 1
6. Item:	7. Description:	8. Part Numb	er:	9. Eligibility:*	10. Quantity:	11. Serial /	Batch No:	12. Status / Work:
001	KIT MOD PRESS	DIFF 914771		TFE731-2-2B	000001	serial not re here i requir tracea HONEYW Refere	OTE: (All ization eferenced sonot be or other by JELL INC. ence FAA 8130.21d)	NEW
13. Remarks: THIS AIRWO		AT 1944 E. SKY HARBO	OR CIRCLE PHOENIX,					
14. Certifies		vere manufactured in conformity to:			3.9 Return to Servi			specified in Block 13
	11.	a and are in a condition for safe ope n data specified in Block 13.	ration.	Certifies that unless of Block 13 was accompanies respect to that work,	olished in accordan	ce with Title	14, Code of Federal B	Block 12 and described in equiations, part 43 and in
15. Authoriz	zed Signature:	J., D.	16. Approval/Authorization N	o: 20. Authorized Signa	ture:		~	21. Approval/Certificate No:
	6	Zinda Fanticki	ODARF602216NM					
17. Name (Typed or Printed):	/	18. Date (m/d/y):	22. Name (Typed or	rrinted):			23. Date (m/d/y):
	LINDA FANTECHI		MAY/05/2005			* 80-7		
			User/Installer	Responsibilities				
It is import	ant to understand that the ex	istence of this document alone does	not automatically constitute auth	ority to install the nart/compo	nent/assembly. W	here the user/	installer performs wo	rk in accordance with the national

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 & 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Serial No: 368 Entered By: MG Otal Time: 8750.8 Total Landings: 6319 Work Order No. Item No. OS-OS Discrepancy: Brow Switch Guards For Left Fuel Ship OFF **INTER CONNECT BROKEN MISSING** Corrective Action: Zeflaced Switch Suggest Automatic Switch Suggest Switch Suggest Switch Swi	egistration:	<u> 3386</u>		Manufa	cture: v	VESTWINE	>	Model N	0: //2	4			
State Time: 8750.8 Total Landings: 6314 Work Order No. Item No. C55-C5 Discrepancy: Brow Switch Guards For LEFT Fuel Show OFF Y INTERCONNECT BROKEN MISSING Corrective Action: Reflect Switch Guards For Table 117. Mile Tech Tile 1500. Corrective Action: Reflect Switch Guards For Table 117. Mile Tech Tile 1500. Corrective Action: Reflect Switch Guards For Table 117. Mile Tech Tile 1500. Sended Oale Reinstalled Date Function CK Date Leak CK Sile 1500. Position Pin Off Sin Off Pin On Sin On Switch Guards Table 117. J. Table 1				Serial N	ت :lo	368							
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herein has been done satisfactorily using the aircrafts airframe, engine, or appliance manufacture's (as applicable) recommended methods, and procedures contained within the manufacture's current maintenance applicable) recommended methods, and procedures contained within the manufacture's current maintenance.	Position switch accordance herein has be	in 255/ 4 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	P/N OF	\$ 43.11-in using the and proc	nsp's) I	N OFF certify that the a airframe, engontained with	work on the in the in	P/N ON 3117- described appliance manufaction	perfor	med, a	S/N ON		
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'rimec Aviation, Inc. 100 AVIATION WAY T. WORTH, TX 76106

/oice: ax:

(817) 626-1376 (817) 626-1402 Packing Slip
Invoice Number:

Invoice Date: May 6, 2005

Page:

Sales Order Number

Drop Shipment

Ship to: STARFLITE MGMT

Sold To: GENERAL DYNAMICS W6355 ATLANTIS DRIVE APPLETON, WI 54914

Customer ID	Customer PO	Payment Terms Net 10 Days			
GD APPLE	A6902141Y				
Sales Rep ID	Shipping Method	Ship Date	Due Date		
	Fed Ex P1	5/6/05	5/16/05		

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	and the second of	may elect to radius a	Switch	<i></i>	L
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Westwind 1124

Pre and Post Flight inspection

8355

Post Fli	ight	
	Nose torque link disconnected	
	Lav serviced	
	Pitot and static covers installed	
	Gear and tires for condition	
	Engine oil levels (1 qt low)	
	Signature	
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Pre Flig		
្នាម ្រាស		
111.0	Windshields and cabin windows for condition	
INI I	Wings, de-ice boots and stall strip for condition	
111111	Tip tank fuel valve closed (up)	
MM	Alterons, flaps and speed brakes for condition	
MALH	HP repture membrane in place	
TANKY	Hyd reservoir level	
TIMA	Gear extension cylinder (1700-2000 psi)	
YM!	Thrust reverser cylinder (710-940 psi)	
1101	Engines for leaks	
11/1	Engir e inlets and exhaust for FOD	
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4 1. 1	nterior systems checked	
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11/12	Fire inflation (nose 55psi main 155 psi)	
	Sump fuel	
	Signature / //	Date 05-11-05
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gistration:	N835G	Manufactu		Model No:	1124	/
ite:	5/26/05	Serial No:	368	Entered By:	DK.	
tal Time:	8762.1	Total Landings	: 6323	Work Order N	0.	item No
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Registration: 1283					Model No: 112P			
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otal Time: 8742		Landings: 6	307	Work Or	der No.	Item !	Vo.	
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applicable) recommend manual. The subseque	ded methods, a	and procedures con	tained within	the manufacture to the work	ure's current performed w	maintena ith the da	nce te on	
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	ns performed i	andlor GMM for the	e required re	oturn to service	statement.	ouren		
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manual, T	ne subsequi	ent signature.	with certificate	number appropr	riate to the work perform work performed and co	ned wit	h the date or
maintenar	ce/inspectio	ons performed	in accordance	with 14 CFR § 1	135.411(a)(2), referenc	e the c	urrent
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in accomplication in application in	ordance with has been do able) recommal. The subscit was compenance/inspecte holders	P/N of the property of the pro	9 (§ 43.11-Insp's) ly using the aircrade, and procedure e, with certificate es a return to sended in accordance ram and/or GMM Cert A&P	I certify that the offs airframe, en es contained with number approprice only for the with 14 CFR §	P/N O e work describ gine, or applia hin the manufa riate to the wo work performe 135.411(a)(2), d return to sen	ed, performance manufal acture's current performed and compared and co	S/ sed, and acture's rent ma ed with t pleted t the curr ent. Date	d comp (as intena the da herein.	ofete nce te or

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inspec	or Signature	1	25-3640			4-27.		

DUTHINGTON AVIOLTION 1.APPROVING NATIONAL AVIATION AUTHORITY/COUNTRY: FAA / UNITED STATES

AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

3.Form Tracking Number: W. 7.F

04 - 64/9 5.Work Order/Contract/Invoice

4.Organization:CONSOLIDATED AIRCRAFT SUPPLY CO. INC. (631) 981-7700 800-422-6300 55 RAYNOR AVENUE, RONKONKOMA, NEW YORK 11779 FAX: (631) 981-7706

Number: R/1/002

	6. Item:	7. Description:	8. Part Number	9. Eligibility:*	10.Quantity:	11.Serial/Batch Number:	12. Status/Work:
I		Hydraulic	713010-505				
	1	Pump	(A3102A)	NA	ONE	12210	OVERHAULED

13. Remarks: 1, THIS UNIT OVERHAULED/REPAIRED/FUNSTIONALLY TESTED IN ACCORDANCE WITH MANUFACTUREF

2. DETAILS OF THE WORK ARE CONTAINED ON ATTACHED WORK ORDER.

3, CEHTIFIES HAT THE WORK SPECIFIED IN BLOCK 1213 WAS CARRIED OUT IN ACCOMMING WITH EASH ART 193 AND WITH RESPECT TO THAT WO

1. I.A.W WELDON A3102A Rev. G. 2-02

14. Certifies the items identified above were manufactured in conformity to:

19. 14 CFR 43.9 Return To Service 4 Other Regulation Specified in Block 13

☐ Approved design data and are in condition for safe operation.
☐ Non-approved design data specified in Block 13.

Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, Part 43 and in respect to that work, the items are approved for return to service

TWA

15. Authorized Signature: 16. Approval/Authorization No:

20. Authorized Signature:

21. Approval/Certificate No: GI 1R167K

17. Name (typed or printed)

22. Name (typed or prir ...)

23. Date (m/d/y): /-22-05

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the co

State Block 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must coorda. In the national regulations by the user/installer before the aircraft may be flown.

18. Date:

n installation certification issued in

CONSOLIDATED AIRCRAFT SUPPLY CO., INC. ₹ 55 RAYNOR AVE. RONKONKOMA, NY 11779

FAA REPAIR STATION #GI1R167K DATE 11-23-04 JOB #04-6419

NOMENCLATURE: Hydrauli	C Piama P	WORK PERFORMED:0VERH	AULED & TESTED
	05 (A3102A)	CUSTOMER: WesThingTon	~ AV.
SERIAL NUMBER: 12210	03 (7)310211 7	CUSTOMER P.O. #: R//	
AMEND./ MOD. PLATE STATUS	: _A /A	MANUFACTURER: WELD	
REQUESTED/REQUIRED MODII YES NO		REVISED PART #: REVISED SERIAL #:	4
AIRWORTHINESS DIRECTIVE SEARCH BY	W.T.F (TECH'S INITIALS) A.D.	FOUND: YES (COMPLETED BLOC	CK 6) NO
1) RECEIVING REMARKS: (CUSTOME OURTH aul.	R INSTRUCTIONS, AIRCRAFT REMC	VALINFO, T.S.O, T.S.N)	
2) TEARDOWN REMARKS: STation Pump Blades, Blade R Cage Seal And Spring	letainer word. Par	lousing worn, not Repair kings, Bearings, Brush	rable. es worn.
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INSPECTION REMARKS: (EXPLAIN	P/N OR S/N CHANGES)		RES NO L

TECHNICIANS SIGNATURE William Front

	o Z		SERVICEABL	E PART
	ACCESSORIES, niff 77061 ax: 713-991-9117 U9BR618Y	STAR FLITE	<u> </u>	20-3113
	51 713-99 R618Y	Nomenclature	Part No. 38451	Serial No.
ith the	AND 62 Bra 111 • F 111 • F S. No.	Work Accomplished Overhauled Other	Bench Checked (Explain)	Repaired
	UNITED BATTERIES 77/ Housig Phone: 713-991-9 F.A.A. C.R.3		MAINTENANCE RELEASE NENT IDENTIFIED ABOVE WAS INSPECTED I AVIATION ADMINISTRATION AND IS APPROVE E AT THIS REPAIR STATION	

UNITED BATTERIES AND ACCESSORIES, INC.

CRS #U9BR618Y 7762 BRANIFF HOUSTON, TX 77061

TEL: 713 991-9111 FAX: 713 991-9117

E-MAIL: unitedbatteries@aol.com

FORM #9007

NiCad Battery Pack Work Order # 8859

Customer Star flite Date Received 4/26/05 Customer PO# N85SG
Battery Pack/Cell/
Power Supply Mfg. Marathon Mfg. Type 20-S113 Mfg. Part No. 38451-004
Serial No. N/A
Pre-Service Hidden Damage Inspection
General Condition PER HANUFACTURERS SPEC Receiving Voltage: Terminals 25.8 Battery Pack#1
Receiving Voltage: Terminals 25.8 Battery Pack#1 72 Pack
Outer Case PHS Circuit Board WA Cells/Cell Cases PHS Hot Spots Now E
Receptacle Assy. 775 5 Amp Fuse 75 10 Amp Fuse 75 25
In-Service Hidden Damage Inspection
Consider Torts, Step 1 through 3
Irst 20.20.20.20 PER CELL AUG.
End of scharge Voltage 24,1
2nd
End of Discharge Voltage
3rd
End of Discharge Voltage
FINAL CHARGE
4th ,54,54,54,54,54,54 DER COEC AUB
End of Charge Terminal Voltage 30,9
Functional Test
Replaced
Remarks
WARRANTY: Months from date of return to service if serviced a minimum of every 6 months.
Inspected By American Completion Date 4-27-05
Inspected By Lawrence Completion Date
Inspector Recommendations The aircraft component, appliance or accessory was inspected in accordance with the current requirements of the Federal Aviation Administration and is approved for return to service. Pertinent details are on file at this repair station. Authorized Inspection Signature Date of Final Inspection for Return to Service Maintenance Release Issued YES or NO

Fly with the Best UNITED

Houston

	No. U9BR618Y	#9016
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	SERVICEABL	
Customer STAR FLITE	-	Make ET PS-823 B/T
Nomenclature	Part No. 501-1075-06	Serial No.
Work Accomplished Overhauled	Bench Checked	Repaired
Other (Explain)	
	MAINTENANCE RELEASE	

CEDVICEADIE DADT

REGULATIONS OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE.

UNDER WORK ORDER NO



UNITE—BAFTERIES AND ACCESSORIES, INC.

CRS #U9Bkul8Y 7762 BRANIFF HOUSTON, TX 77061

TEL: 713 991-9111 FAX: 713 991-9117

E-MAIL: unitedbatteries@aol.com FORM #9007

NiCad Battery Pack Work Order # 8860

Customer Star flite Date Received 4/26/05 Customer PO# N83SG
Battery Pack/Cell/
Power Supply Mfg. Jet Mfg. Type PS-823 B/T Mfg. Part No. 501-1075-06
Serial No. <u>1310</u>
Pre-Service Hidden Damage Inspection
General Condition PER HANUFACTURERS SPEC
Receiving Voltage: Terminals 25,4 Battery Pack#1 25,4 ½ Pack
Battery Pack #2 25.4
Outer Case PHS Circuit Board PHS Cells/Cell Cases PHS Hot Spots NOWE
Receptacle Assy. Phs 5 Amp Fuse Phs 10 Amp Fuse Phs
In-Service Hidden Damage Inspection
Capacity Tests, Step 1 through 3
11st ,08,08,08,08,08 - PER COUL NO.
End of Discharge Voltage 21,66
2nd
End of Discharge Voltage
3rd
End of Discharge Voltage
FINAL CHARGE
4th ,55,55,55,55,55 PER RECLAUG
End of Charge Terminal Voltage 3/1/
Functional Test PHS PER CHH
Replaced
Remarks
WARRANTY: Months from date of return to service if serviced a minimum of every 6 months.
Work Accomplished By Milan Stede per CMM TP-202
Work Accomplished by
Inspected By Your Kan Completion Date 120-05
hispected by <u>years</u> Completion Date
Inspector Recommendations
The aircraft component, appliance or accessory was inspected in accordance with the current requirements of the
Federal Aviation Administration and is approved for return to service.
Pertinent details are on file at this repair station.
Anthorized Ingression Signature
Date of Final Inspection for Return Serve Maintenance Release Issued YES or NO
Date of I that improved to I term,





Registration: NS'	<u>356</u>	Manufacture:		Model No:	1124	R
Date: 3-1-0		Serial No:	368	Entered B	y: ORS	
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manual. The subsequent	signature, with certificate	number appropriate	e to the work performe	ed with the date of
	constitutes a return to service performed in accordance			
	red program and/or GMM,			
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gistration: $N8$	356	Manufacture Serial No:	368	Model N Entered			
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tel Time:	8733.0	Total Landi	ngs:	630 J	Work Order	No.	ltem	No.
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General Maintenance Manual

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herein has been done satisfa applicable) recommended m	ethods, and procedures	s airframe, engir contained within	e, or appliance mathematical threats	s current	maintenar	nce a on
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General Maintenance Manual

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	oving National Aviation Aviation hority/Country:	AUTHORIZED RELEA	SE CERTIF	ICATE	3. Form Tracking Nu	
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- 6. nem:		8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
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CUS Dis	screpancy: Au	ITE MANAGEMENT GROUP topilot oscillates in r dings: Bench checked ur ound unit to be in serv	nit and coul	ld not du	ıplicate	
Cor	mputer Instru CHNICIAN:ANDR					ilot
14. C	Approved design d	above were manufactured in conformity to: ata and are in a condition for safe operation gn data specified in Block 13.	Certifies that unle in Block 12 and d	ess otherwise spe lescribed in Bloc de of Federal Re	ce Other regulation cified in Block 13, the half was accomplished gulations, part 43 and for return to service	ne work identified ned in accordance and in respect to

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

16. ApprovalAuthorization No.:

18. Date (m/d/y):

the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in , it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts parts/components/assemblies from the airworthiness authority of the country accepts accepts and accept accepts and accept accepts a seminary accepts and accept accepts accepts a seminary accepts

User/Installer Responsibilites

20. Authorized Signature:

22. Name (Typed or Printed): STEVE J. KRINGS

Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

15. Authorized Signature:

17. Name (Typed or Printed):

21. Approval/Certificate No.:

23. Date (m/d/y): 2/02/05

JGVR194F

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TEC	CHNICIAN: AND	REW B	ERG				
	This docum	ent c	onstitutes a signed	l copy of the t	work c	order.	
14. Ce	rtifies the items identified	l above we	re manufactured in conformity to:	19. X 14 CFR 43.9 Retu		_	specified in Block 13
	•		are in a condition for safe operation. specified in Block 13.	Certifies that unless othe in Block 12 and describe with Title 14, Code of F that work, the items are	ed in Block ederal Reg	t 13 was accomplished	ed in accordance I in respect to
15 Au	thorized Signature:		16. ApprovalAuthorization No.:	20. Authorized Signature:	,	DUNCAN 21. Appro	val/Certificate No.:

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

16. ApprovalAuthorization No.:

18. Date (m/d/y):

the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in , it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

User/Installer Responsibilites

Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

20. Authorized Signature:

22. Name (Typed or Printed): AARON J. SPULAK

JGVR194F

23. Date (m/d/y): 3/07/05

15. Authorized Signature:

17. Name (Typed or Printed):

3. Form Tracking Number: 2. 1. Approving National Aviation Authority/Country: AUTHORIZED RELEASE CERTIFICATE NY8KC0001001 FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG FAA/UNITED STATES 5. Work Order/Contract/Invoice 4. Organization Name and Address: Number: NY8KC Duncan Aviation/Lincoln Airport/Lincoln, NE 68524 12. Status/Work: 10. Quantity: 11. Serial/Batch Number: Eligibility: * 8. Description: Part Number: REPAIRED 001 1950 N/A622-3108-016 001 AUTOPILOT AMPLIFIER 13. Remarks: CUSTOMER: STARFLITE MANAGEMENT GROUP Discrepancy: Autopilot oscillates in roll. Preliminary Findings: Bench checked unit and found roll rate cutout circuit defective. Corrective Actions: Troubleshot unit and replaced defective components in Function tested unit per Collins APA-80() Autopilot roll rate circuit. Amplifier Repair Manual. TECHNICIAN: ANDREW BERG This document constitutes a signed copy of the work order. 19. X 14 CFR 43.9 Return to Service Other regulation specified in Block 13 14. Certifies the items identified above were manufactured in conformity to: Certifies that unless otherwise specified in Block 13, the work identified Approved design data and are in a condition for safe operation. in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to Non-approved design data specified in Block 13. that work, the items are approved for return to service. $\overline{D}_{u_{B_{\psi\eta_{B}}}}$ 21. Approval/Certificate No.: 20. Authorized Signature: 16. ApprovalAuthorization No.: QI_{43} 15. Authorized Signature: JGVR194F 23. Date (m/d/y): 22. Name (Typed or Printed): 18. Date (m/d/y): 17. Name (Typed or Printed): 2/02/05 KRINGS STEVE J. User/Installer Responsibilites It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. he user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in its essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown. NSN:0052-00-012-9005

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STARFLITE AVIATION DAILY AIRCRAFT & ENGINE LOG

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LOG# 2272

STARFLITE AVIATION DAILY AIRCRAFT & ENGINE LOG

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Total - In adding the second			w.u							F	AR 91.4	111			
WITH RESP	ECT TO THE WO	RK PERFORMI	ED ABOVE, TH	S AIRCRAFT I	S APPROVED I	FOR RETURN	TO SERVICE.			F	AR 91.4	113			
SIGNATI	JRE:			A & P	#					F	AR 135.	185			i i i i i i i i i i i i i i i i i i i
THIS TR	IP CTED FOR:	K	OGER			· · · · · · · · · · · · · · · · · · ·				FU	EL PUR	CHASE L	OG		
THIS AIRCR	AFT HAS BEEN	NSPECTED IN	ACCORDANCE	WITH AN FAA	APPROVED A	FM PREFLIGH		GALLONS	777	, <u>S</u>	ARFL17	E S		CC TYPE	
SIGNATI		well	·	CERT	<u>#398</u>	3021	38	GALLONS	150	LOC	FTW	\$		CC TYPE	Anex
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Date: 1-27-05 Serial No: 348 Entered By: DAS Total Time: 8721.6 Total Landings: U29/ Work Order No. Item No. 02-05 Discrepancy: Cff From 01-05-9 Clw 54.2492 Hyd. Hose Insp. Duce 9749.7 Corrective Action: iwp. hyd. sys. flexible have no defects noted Tech Time	Registration: 23	56	Manufacture:		Model No:	1124
Discrepancy: CFF From 01.05-9 Cliw 54.2492 Hyd. Hose Insp. Duel 9749.7 Corrective Action: Imp. Byd. 195 Flexible hase no perfects notice Tech lines Removed Date Reinstalled Date Function CK Date Leak CK Date Position P/N OFF S/N OFF P/N ON S/N ON In accordance with 14 CFR § 43.9 (§ 43.11-Insp's) I certify that the work described, performed, and completed herein has been done satisfactorily using the aircrafts airframe, engine, or appliance manufacture's (as applicable) recommended methods, and procedures contained within the manufacture's current maintenance manual. The subsequent signature, with certificate number appropriate to the work performed with the date on which it was completed, constitutes a return to service only for the work performed and completed herein. For maintenance/inspections performed in accordance with 14 CFR § 135.411(a)2), reference the current certificate holders approved program and/or GMM, for the required return to service statement. Technician Signature Centificate Number Date Date Date Centificate Number Date	Date: 1-27	-05	Serial No:	348	Entered By:	Dns
Corrective Action: IMP hyd. sys. Flexible have NO DEFECTS NOTED Tech Time Removed Date Reinstalled Date Function CK Date Leak CK Date Position P/N OFF S/N OFF P/N ON S/N ON In accordance with 14 CFR § 43.9 (§ 43.11-Insp's) I certify that the work described, performed, and completed herein has been done satisfactorily using the aircrafts airframe, engine, or applicance manufacture's (as applicable) recommendent signature, with certificate number appropriate to the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed, constitutes a return to service only for the work performed with the date on which it was completed. Removed Date Leak CK Date Times Ti	Total Time: 🕺 🧎	21.6 To	tal Landings:	4291	Work Order N	o. Item No.
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San Juni A&P 2668256 2-16-05 Inspector Signature Certificate Number Date	Certificate noiders a	ipproved progra Signature	Certif	icate Number	REMITTE SERVICE STATE	
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1/ 1/ 1 ARP 253660856 2-16-05	Inspector S	ionature	Certif	icate Number		Date
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Registration:	835	- I no.	anufacture:	TAL	Model No	112	4	
	7.05		erial No:	368	Entered I		5	
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maintenance/i	nspections pe	erformed in a	ccordance w	ith 14 CFR § 13!	5.411(a)(2), refe	rence the c	urrent	
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Insper	ctor Signature			cate Number		Date		
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Registration: 🦎		Manufacture:	IAT	Model No:	1124
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Registration: N (356-	Manufacture:	TA4	Model No:	1124
Date: 1-17.01		Serial No:	346	Entered By:	DRS
Total Time: 872	1.6 Tota	I Landings:	6291	Work Order N	o. Item No.
				02.05	4
Discrepancy:	NILL	two wh	oer Blade	· Insorts	on Odor.
Discrepancy:					
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Corrective Action:	Nstalled .	2 NEW Wiper	blades PN	2315M-24-	
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Removed Date					
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Registration: $\sqrt{3}$ Date: $\sqrt{2}$ Total Time: $\sqrt{8}$ Discrepancy: Corrective Action:	Line To	Serial No: tal Landings: Light In	348 4291 00.	Work Order	No.	Item I	No.
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				c Kgwc/		Tech	
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corrective Action:	replaced b	ull PN DA-	27 Ops che	i Kgwel		Test	
orrective Action:	replaced be	ell on DA-	27 Ops che	ckgod		Tool	
corrective Action:	replaced b	ill on DA-	27 Ops ch	ekgood		Tech	-
orrective Action:	replaced be	uls vi VH-	al Ops che	-cKgood		1100	7.
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Position	P/N OF	F S/I	OFF	P/N ON		S/N ON	

Registration: 🛠	3 S G-	Manufacture:	IAI	Model No:	1124		
Date: 1-27,		Serial No: 3	G 8	Entered B	y: DA	S	·
Total Time:	To	otal Landings:		Work Orde	r No.	item i	No.
				02-05		ý	
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Discrepancy: K.	1. P 14/	CK TATA CONE	STAME.	51r.p 60	rroded		<u> </u>
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Corrective Action:						Tech	Time
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In accordance with	14 CFR § 43.9	(§ 43.11-insp's) i ce	ertify that the wo	rk described, pe	rformed, a	nd comp	oleted
		using the aircrafts a					noe
manual. The subse	quent signature.	with certificate nun	ber appropriate	to the work per	formed with	h the dat	e on
which it was comple	eted, constitutes	a return to service	only for the worl	k performed and 411/a)/2) refere	completed	d herein. Irrent	For
certificate holders a	pproved progra	m and/or GMM, for	the required retu	urn to service st	atement.		· .
<u>Technician</u>	Signature	Certifica	te Number		Date		
		A&P					
Inspector S	<u>ignature</u>	Certifica	te Number		Date		
		A&P					

	356	Manufacture:	IAT	Model No:	1124
Date: 1-27	1.05	Serial No:	368	Entered By:	ONS
Total Time: 87	21.6 To	tal Landings:	6291	Work Order No	o. Item No.
				02.05	7
Discrepancy:					
	+ Re-	FILL O	KURIN S.	stem	
			/		
Corrective Action:	REFILL	ED OXYGEN	Sween	TO 1850	Tech Time
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which it was comple maintenance/inspec	tions performed	I in accordance with	14 CFR § 135.41	1(a)(2), reference	e the current
certificate holders a	pproved prograi	m and/or GMM, for	the required returnite Number	n to service state	ment. Date
Technician S	Killamic	2000	1.0851 /2-	a 4	
Inspector S	ionature	A&P S (0)	ate Number	d-1	Date
- madecial G	-Armsha				
		A&P			

General Maintenance Manual

	N3656 16-05	Manufacture: Serial No:	<u>taI</u> 348	Model No: // 2 Entered By:	DM DM	
Total Time:	872116	Total Landings:	6291	Work Order No.	Item N	0.
				02-05	8	
Discrepancy:	CIW	A.D 2004-1	13-20 G	02-05 BRM!n 605		
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			GM	0		يدانن
Corrective A	ction: A.D. listed units	2004-13-20 IN	NA this ac	Fl. Not equiped	Tech 1	line .5
					Q.	.5
						
						
						
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				rk described, performed		
herein has be applicable) re manual. The which it was a maintenance certificate ho	een done satisfacti scommended meti subsequent signa completed, constit /inspections perfo	orily using the aircraft hods, and procedures ture, with certificate nutes a return to service med in accordance wogram and/or GMM, for Certificate A&P	s airframe, engine, contained within the umber appropriate se only for the work with 14 CFR § 135.4	or appliance manufacture manufacture's current to the work performed via performed and comple 111(a)(2), reference the urn to service statement	ire's (as t maintenant with the date ted herein. F current	ce on
				~ /1- /		
Mua	4/1 Dest.	A&P 459	981960	2/1/2/	05	

Registration: $\sqrt{3}$		Manufacture: IA Serial No: 368				
					14	
Total Time: Ø	8/2/16	Fotal Landings: 1/2	4/ Work	Order No.	Item	No.
			02-4		9	
Discrepancy:	INSpec	t under water	Locator De	escen Ba	HOS	V
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		DUKONP	· Jyr			
					No.	
Corrective Action:	cfw	under water Locator NEFECTS NOTED	beaucon battery	inen.	Tech	Tine
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In accordance with	MICERIS 431	(§ 43.11-insp's) I certify	hat the work describe	o operanne	and com	Jeter
herein has been do	ne satisfactoril	y using the aircrafts airfrar	ne, engine, or appliar	ce manufactur	e's (as	
applicable) recomming manual. The subse	nended method	is, and procedures contain s, with certificate number a	ed within the manufactors	cture's current	maintena th the de	nce
which it was comple	eted, constitute	es a return to service only	or the work performe	d and complete	ed herein.	For
maintenance/inspe	ctions performe	ed in accordance with 14 (am and/or GMM, for the re	FR § 135.411(a)(2), i	reference the c	urrent	٠,
Technician		Certificate Nu		<u>Date</u>		
den	Lynch	A&P 2669256		2-16-05		
Inspector \$	/	Certificate Nu		Date		
1 (ama)4	14 L	ABP 459881966		2/10/6	\1 -	
MUMMY 1	May	737101100	<u> </u>	21,010		

STARFLITE AVIATION General Maintenance Manual

		nufacture: IAI rial No: マレグ	Model No: Entered By:	1124
otal Time: 872	-1,4 Total La	andings: 6291	Work Order	
			02.05	10
iscrepancy:	Clw A.O.	2003-26-14	· HALON EI	10 BOHC
			yo.	
orrective Action:	c/w AD 2003	3-26-14 - Removal F	ire bottles & INST	talked Tech Tie
2 New Fir	re bottles & mt.	brackets		\$ 1.
				Oh.
Removed Dak	e Reinstalled	Date Function CK	Dale	eak CK Date
Position	P/N OFF	S/N OFF	PINON	SINON
cabin	898052	V- 329659	A352	1-527017
cockpit	898052	V-329516	A352	V-130195
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		the aircrafts airframe, eng	ine, or appliance mai	nufacture's (as
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erein has been done pplicable) recomme	ended methods, and pent signature, with o			
erein has been done pplicable) recomme nanual. The subsequence it was complete	uent signature, with d ed, constitutes a retu	pertificate number appropri urn to service only for the v	ate to the work performed and controls and controls are to the control are the control are to the control are	rmed with the date of completed herein. For
erein has been done applicable) recomme nanual. The subsequal which it was complete naintenance/inspect	uent signature, with ded, constitutes a retuions performed in ac	certificate number appropri arn to service only for the vicordance with 14 CFR § 1	ate to the work perfo vork performed and o 35.411(a)(2), referen	rmed with the date of completed herein. For completed herein, For contract the current
erein has been done applicable) recomme nanual. The subsequal which it was complete naintenance/inspect	uent signature, with ded, constitutes a retuions performed in acoproved program and	pertificate number appropri urn to service only for the v	ate to the work perfo vork performed and o 35.411(a)(2), referen	rmed with the date of completed herein. For completed herein, For contract the current
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nerein has been done applicable) recomme manual. The subseque which it was complete maintenance/inspectional process applied to the control of the control o	uent signature, with ded, constitutes a retuions performed in acoproved program and tonature A&	certificate number appropri urn to service only for the v cordance with 14 CFR § 1 l/or GMM, for the required	ate to the work perfovork performed and commendations and commendations are the service states.	rmed with the date of completed herein. For ce the current ement. Date

Registration: 1/3 1/5 6- Date: 2.11.15	Manufacture: Serial No:	IAI 348	Model No: 1/2 Entered By: Ox	
Total Time: 87216	Total Landings:	4291	Work Order No.	Item No.
			0205	11
Discrepancy: C/w	VISUAL F	ind + st	0205 Portable FI	~
So H	<u> </u>			
Corrective Action:	lw find g aft. Fi	re both insp	NO DEFECTS NOTE	o Tech Time
				13 .5
Removed Sale	Reinstalled Date	Function CK	Date Leak CK	Date
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			PAR-OR	GAT ON
In accordance with 14 CPR § herein has been done satisfa applicable) recommended m	ctorily using the aircrafts a ethods, and procedures co	airframe, engine, ontained within th	or appliance manufactue manufacturers	ire's (as i maintenance
manual. The subsequent sign which it was completed, con- maintenance/inspections per certificate holders approved	stitutes a return to service formed in accordance with	only for the work	performed and comple 111(a)(2), reference the	ted herein. For current
Technician Signature Sea Sun!	A&P 2668	ate Number	2-16-0	
Inspector Signature		ate Number	Da	
Nauil H Den	6. A&P 45996	1960	2/16/01	

Total Lan	+ has		Entered By Work Order 2-05 - wite on	No.	item 1 12 118	
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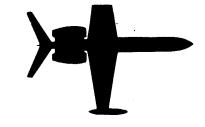
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STARFLITE AVIATION General Maintenance Manual

Date: 02-13-2005 Serial No: 398 Entered By: 1/2 Total Time: 872/14 Total Landings: 429/ Work Order No. 02-05 Discrepancy: 0xy 6EN 00 W Corrective Action: SERVECED 0xy 6EN Removed Date Reinstalled Date Function CK Date Leak CK	Item No.
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Corrective Action: SERVECED OXYGEN	Tech la
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	Date
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	and the second

Registration: N8356 Date: 02-17-2005		Manufacture: プタエ Serial No: 368		124A
Total Time: \$721.6	Total Landings:	6291	Work Order No	
			02.01	14
Discrepancy: PILOT R INCREASED TO	EQUEST TO	RE PRE	SSURE UND RIGH	BE II MAENS
Corrective Action: SERVI	CED LEFT	AND RI	SAI MAT	~ Tech Time
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In accordance with 14 CPR § 43 herein has been done satisfacto applicable) recommended method manual. The subsequent signate which it was completed, constitutional maintenance/inspections perform certificate holders approved protechnician Signature	rily using the aircrafts and procedures of ure, with certificate nurities a return to service med in accordance with gram and/or GMM, for	airframe, engine, ontained within the nber appropriate only for the work in 14 CFR § 135.4	or appliance manu- ie manufacture's cu to the work perform performed and co i11(a)(2), reference	facture's (as urrent maintenance ned with the date on mpleted herein. For a the current
Samy P. Dam	1 A&P 2718	182	02-17.	
Damil / Sent	A&P 4599	SIGUS	2/17/	Date 8 S

MZ4-WEJTHINU



SERVICE LETTER

SERVICE LETTER NO. WW-2492

DECEMBER 31, 1982

EFFECTIVITY:

1124/1124A WESTWINDS

SUBJECT:

HYDRAULIC HOSE INSPECTION

COMPLIANCE:

AT EACH SCHEDULED INSPECTION

APPROVAL:

IAI ENGINEERING

REASON:

TO REMIND MAINTENANCE PERSONNEL OF THE LEAK INSPECTION REQUIREMENTS FOR ALL HYDRAULIC SYSTEM HOSES AND TO PAY SPECIAL ATTENTION TO THOSE THAT CONTAIN A RUBBER LINER MATERIAL (COLOR CODED BY A GREEN EXTERNAL

APPEARANCE).

INSTRUCTIONS:

A routine inspection should be performed of all hydraulic system flexible hoses for leakage or deterioration in accordance with the requirements of Chapter 5 or the 50 Hour Phase Inspection Program. The materials used in teflon lined hoses are more resistant to wear and deterioration than the rubber lined hoses. Either type hose may be used for replacement of defective hoses, however, the teflon lined hoses should provide extended service life.

SUPPLY DATA:

Replacement hoses may be obtained from Atlantic Aviation Supply Co. or their dealers. See 1124/1124A Illustrated Parts Catalog for hose part numbers and quantities.

WEIGHT AND BALANCE:

N.A.

AIRCRAFT RECORDS:

No recurrent entry in the airplane logs will be required, however it may be desirable to record lines which have been changed to teflon lined material.

		UTHORIZED RELEA Form 8130-3, AIRWORT			3. Form Tracking Num 611023 Page 1 of 1	nber:
l. Organi	zation Name and Address: Rosemount Aer 14300 Judicial F Burnsville, MN	Road	FAA-PMA and FAA-TS approval holder	0	5. Work Order/Contra 49756	
. Item	7. Description	8. Part Number	9. Eligibility*	10. Quantity 11	Serial/Batch Number	12. Status/Work
1	BLADE REFILL	2315M-24-8	IAI 1123 IAI 1124 IAI 1125 WESTWIND ASTRA IAI ASTRA SPX	22	NA	New
for F	WORTHINESS APPROVAL-PARTS. THIS F PMA or TSO authorizations.			. as indicated in block	3	
	ife parts must be accompanied by maintenance h		time since new.			
14. Certi	fies the items identified above were manufactured Approved design data and are in a condition to the second secon	or safe operation	19. 14 CFR 43.9 Retu Certifies that unless specifies was accomplished in account to that work, the items are	ied in block 13, the work rdance with Title 14, Cod	identified in Block 12 and o e of Federal Regulations, p	ion specified in Block 13 described in Block 13 part 43 and in respect
15. Auth	norized Signature	16. Approval/Authorization No.: DMIR-501780-CF	20. Authorized Signature:			21. Approval/Certificate No.;
17 Nam	ne (Typed or Printed): Jack A. Benson	18. Date (m/d/y): / / / / / / / / / / / / / / / / / / /	22. Name (Typed or Printed	1):		23. Date (m/d/y)
			aller Responsibilities			
Where parts/co	contant to understand that the existence of this Document alone do the user/installer works in accordance with the national regulations omponents/assemblies from the Airworthiness Authority of the cou- ents in 14 and 15 do not constitute installation certification. In all c	s of an Airworthiness Authority different than the A ntry specified in block 1.	irworthiness Authority of the country specified			

NSN: 0052-00-012-9005

Shoreline dba Safetech Houston Branch - FAA WV1R599K 8305 Monroe Road Houston, TX 77061 713-947-1115 713-947-0593 (fax)

Work Order No: MH11227

Rec'd: 01/26/2005

Date Shipped: //

Date Reqd: 02/04/2005

Customer: STARFLITE MANAGEMENT GROUP

Cust # 895

Account Code: CREDIT CARD

Customer P.O. # 608MD

Instructions: HYDRO

MFR: AMEREX

Part Description: HAND HELD EXTINGUISHER

Qty: 1

MFR.#: A352

Serial No: V-130195

Additional Serial No:

Teardown Finding: LOW WEIGHT & LOW PRESSURE, HYDROTEST DUE, MISSING TAMPER SEAL.

Corrective Action: CLEANED AND INSPECTED, HYDROTESTED IAW 49 CFR 180.205, OVERHAULED, FILLED, LEAK CHECKED,

WEIGHT CHECKED (5 LBS. 0 OZ.), INSTALLED TAMPER SEAL, PER MFR MANUAL#05604,

PER NFPA PAMPHLET'S # 10 & 408, PER ADVISORY CIRCULAR AC 20-42C.

Part Number	Description	Qty Order	Qty Issue	Price	Amount
INSPECTION	CLEAN & INSPECT	1	1		
HYDRO-1211	HYDROTEST 1211 FIREX	1	1		
MFLAGBL	BLUE TAMPER SEALPLASTIC	1	1		
5241A	NECK SEAL-AMEREX (344,352,354,355,358)	1	1		
6092A	VALVE STEM - AMEREX(344,352,354,355,358)	1	1		
1211	HALON 1211 (CBRCIF2) - LBS.	2.5	2.5		
LABOR-1211	LABOR FOR 1211 HANDHELD FIRE EXTINGUISHERS	1	1		
UN1044	FIRE EXTINGUISHERS / 2.2 / UN1044 /	1	1		
	NONFLAMMABLE GAS				

Da. 3 Completed: 01/31/2005

Tech: DM / AD

inspector: ERIC KARSTEN

CUSTOMER COPY



Safetech

8305 MONROE HOUSTON, TEXAS 77061 713-947-1115 FAX 713-947-0593

CERTIFICATE OF CONFORMANCE FOR HANDHELD FIRE EXTINGUISHER

FAA REPAIR STATION # WV1R599K

02/03/05 STARFLIGHT HOUSTON,TX

Purchase order: 608MD

Our sales order number: MH11227

It is hereby certified that the article/articles listed below in the quantities as called on the above purchase order number are in conformance with the requirements and specifications of the purchase order and any other requirements as documented on this certification.

Other: CFR 49

NFPA 10 NFPA 408

Advisory Circular - AC 20-42C MFR MANUAL # 05604

Unit has been inspected and / or overhauled IAW the manufacture's manual and the NFPA maintenance requirements.

Last hydrotest: 1/05 Next hydrotest: 1/17 Cylinder mfd: 1/92 Cylinder exp: N/A 6 YR MAINT DUE: 1/11

Quantity: 1

Part Number: A352

Serial Number: V-130195

Quality Assirance Manager ERIC KARSTEN Shoreline dba Safetech Houston Branch - FAA WV1R599K 8305 Monroe Road Houston, TX 77061 713-947-1115 713-947-0593 (fax)

Fwd

Work Order No: MH11226

Date Rec'd: 01/26/2005 Date Shipped: // Date Reqd: 02/04/2005

Customer: STARFLITE MANAGEMENT GROUP Cust # 895

Account Code: CREDIT CARD Customer P.O. # 608MD

Instructions: HYDRO MFR: AMEREX

Part Description: HAND HELD EXTINGUISHER Qty: 1 MFR.#: A352

Serial No: V-527017 **Additional Serial No:**

Teardown Finding: HYDROTEST DUE, LOW PRESSURE.

Corrective Action: CLEANED AND INSPECTED, HYDROTESTED IAW 49 CFR 180.205, OVERHAULED, FILLED, LEAK CHECKED,

WEIGHT CHECKED (5 LBS. 0 OZ.), INSTALED TAMPER SEAL, PER MFR MANUAL 05604, PER NFPA

PAMPHLETS # 10 & 408, PER ADVISORY CIRCULAR AC 20-42C.

Part Number	Description	Qty Order	Qty Issue	Price	Amount
INSPECTION	CLEAN & INSPECT	1	1		
HYDRO-1211	HYDROTEST 1211 FIREX	1	1		
MFLAGBL	BLUE TAMPER SEAL-PLASTIC	1	1		
5241A	NECK SEAL-AMEREX (344,352,354,355,358)	1	1		
6092A	VALVE STEM - AMEREX(344,352,354,355,358)	1	1		
1211	HALON 1211 (CBRCIF2) - LBS.	2.5	2.5		
LABOR-1211	LABOR FOR 1211 HANDHELD FIRE EXTINGUISHERS	1	1		
UN1044	FIRE EXTINGUISHERS / 2.2 / UN1044 / NONFLAMMABLE GAS	1	1		

Tech: DM / AD . Completed: 01/31/2005

Inspector: ERIC KARSTEN

CUSTOMER COPY



Safetech

8305 MONROE HOUSTON, TEXAS 77061 713-947-1115 FAX 713-947-0593

CERTIFICATE OF CONFORMANCE FOR HANDHELD FIRE EXTINGUISHER

FAA REPAIR STATION # WV1R599K

01/31/05 STARFLIGHT HOUSTON,TX

Purchase order: 608MD

Our sales order number: MH11226

It is hereby certified that the article/articles listed below in the quantities as called on the above purchase order number are in conformance with the requirements and specifications of the purchase order and any other requirements as documented on this certification.

Other: CFR 49

NFPA 10 NFPA 408

Advisory Circular - AC 20-42C MFR MANUAL # 05604

Unit has been inspected and / or overhauled IAW the manufacture's manual and the NFPA maintenance requirements.

Last hydrotest: 1/05
Next hydrotest: 1/17
Cylinder mfd: 1/96
Cylinder exp: N/A

6 YR MAINT DUE: 1/11

Quantity: 1

Part Number: A352

Serial Number: V-527017

Quality Assurance Manager ERIC KARSTEN



Airworthiness Directive

/ Federal Register Information

Header InformationDEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-CE-19-AD; Amendment 39-13413; AD **2003-26-14**]

RIN 2120-AA64

Airworthiness Directives; Kidde Aerospace Part Number (P/N) 898052 Hand-Held Halon Fire Extinguishers PDF Copy (If Available):



Preamble Information

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA adopts a new airworthiness directive (AD) for certain Kidde Aerospace P/N 898052 hand-held halon fire extinguishers that are utilized on aircraft. This AD requires you to remove the affected fire extinguishers from service and would prevent you from using them in the future. This AD is the result of information that shows that the discharge time of the affected fire extinguishers exceeds the maximum allowable discharge time. The problem is due to incomplete crimping of the siphon tube. We are issuing this AD to remove from service fire extinguishers that had this incomplete crimping of the siphon tube. If not removed from service, these fire extinguishers could function at diminished levels and compromise the level of safety in an emergency situation.

DATES: This AD becomes effective on February 20, 2004.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulation as of February 20, 2004.

ADDRESSES: You may get the service information identified in this AD from Kidde Aerospace, Kidde Technologies,

Inc., 4200 Airport Drive, NW., Wilson, North Carolina 27896; telephone: (252) 237-7004.

may view the AD docket at FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 223-CE-19-AD, 901 Locust, Room 506, Kansas City, Missouri 64106. Office hours are 8 a.m. to 4 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Charles H. Bowser, Flight Test Engineer, FAA, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, Suite 450, Atlanta, Georgia 30349; telephone: (770) 703-6047; facsimile: (770) 703-6097.

SUPPLEMENTARY INFORMATION:

Discussion

What Events Have Caused This AD?

The FAA has received information of problems with certain Kidde Aerospace P/N 898052 hand-held halon fire extinguishers that are utilized on aircraft. This information shows that the discharge time of the affected fire extinguishers exceeds the maximum allowable discharge time.

The problem is due to incomplete crimping of the siphon tube. Specifically, worn crimping tools were used to crimp the siphon tube. This is causing leakage between the siphon tube and the valve.

rt Is the Potential Impact if FAA Took No Action?

If these fire extinguishers that had this incomplete crimping of the siphon tube are not removed from service, then the fire extinguishers could function at diminished levels and compromise the level of safety in an emergency situation.

Has FAA Taken Any Action to This Point?

We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply certain Kidde Aerospace P/N 898052 hand-held halon fire extinguishers that are utilized on aircraft. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on May 13, 2003 (68 FR 25543). The NPRM proposed to require you to remove the affected fire extinguishers from service and would prevent you from using any affected fire extinguisher in the future.

Comments

Was the Public Invited To Comment?

We provided the public the opportunity to participate in the development of this AD. The following presents the comments received on the proposal and FAA's response to each comment:

Comment Issue No. 1: Extend the Compliance Time

What Is the Commenter's Concern?

Several commenters recommend extending the compliance time from 6 months to 12 months, while one commenter recommends an extension to 18 months. The commenters state that the extension is necessary due to the large number

of affected extinguishers and the logistics involved with AD compliance.

V Is FAA's Response to the Concern?

The FAA agrees that 12 months would be a more realistic compliance time.

We are changing the final rule AD action accordingly.

Comment Issue No. 2: Clarify the Fire Extinguisher Applicability

What Is the Commenter's Concern?

Several commenters state that the current wording for the fire extinguisher applicability of "manufactured from 1995 through 2002 and have a serial number of W-389653 or lower" is confusing. The commenters recommend the following language to more fully depict the intended applicability:

Fire extinguishers affected by this AD are serial numbers V-432001 through W-389653 inclusive that were manufactured sometime from 1995-2002. Serial numbers are identified by the Underwriter's Laboratories (UL) number printed on the label and are listed in succession. Other variants of the UL number with prefixes other than "V" or "W" are not affected by this AD.

What Is FAA's Response to the Concern?

The FAA concurs that the recommended language more accurately reflects the fire extinguisher serial number range.

Ware changing the final rule AD action accordingly.

Comment Issue No. 3: Add a Dash Number to the Existing Part Number

What Is the Commenter's Concern?

One commenter recommends adding a dash number to the existing fire extinguisher part number. The commenter states that this would allow you to distinguish between pre- and post-bulletin modifications.

What Is FAA's Response to the Concern?

The FAA does not believe that this is necessary since the replacement fire extinguishers will have their own separate and unique serial numbers.

We are not making any changes to the final rule AD action.

Comment Issue No. 4: Cost Estimate Too High

What Is the Commenter's Concern?

One commenter states that FAA's estimate of 2 workhours to locate, access, pack, ship, receive the new unit, store, and reinstall the new unit is too high. The commenter states that 1 workhour is a conservative estimate.

V Is FAA's Response to the Concern?

The FAA agrees that 1 workhour more adequately reflects the time necessary to do the work.

We are changing the final rule AD action accordingly.

Comment Issue No. 5: Revise Fire Extinguisher Return Procedures

What Is the Commenter's Concern?

One commenter recommends that the AD should more clearly reference the procedures in the service information for returning any fire extinguishers. Specifically, the commenter states that you should not discharge the fire extinguishers, and you should not ship them back to Kidde because a special collection point is already established. This information is outlined in the service information.

What Is FAA's Response to the Concern?

The FAA agrees that the return procedures should reference that in the service information.

We are changing the final rule AD action accordingly.

Conclusion

What Is FAA's Final Determination on This Issue?

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed except for the changes discussed above and minor editorial corrections. We have determined that these changes and minor corrections:

- -- rovide the intent that was proposed in the NPRM for correcting the unsafe condition; and
- --Do not add any additional burden upon the public than was already proposed in the NPRM.

Changes to 14 CFR Part 39--Effect on the AD

How Does the Revision to 14 CFR Part 39 Affect This AD?

On July 10, 2002, the FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the FAA's AD system. This regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Costs of Compliance

How Many Airplanes Does This AD Impact?

We estimate that this AD affects 38,695 airplanes in the U.S. registry.

What Is the Cost Impact of This AD on Owners/Operators of the Affected Airplanes?

We estimate the following costs to remove the affected fire extinguishers from service (including replacing with ε per unit):

Labor cost	Parts cost	Total cost per airplane

1 workhour X 60 per hour = 60.

No cost for parts.

\$60 per airplane.

ipliance Time of This AD

What Will Be the Compliance Time of This AD?

The compliance time of this AD will be "within the next 12 months after February 20, 2004 (the effective date of this AD)."

Why Is This Compliance Time Presented in Calendar Time Instead of Hours Time-in-Service (TIS)?

Although the slow discharge of the fire extinguishers is only a problem during flight, the unsafe condition is not a result of aircraft operation. Therefore, FAA has determined that a compliance based on calendar time should be utilized in this AD in order to ensure that the unsafe condition is addressed on all aircraft in a reasonable time period.

Regulatory Findings

Will This AD Impact Various Entities?

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

Will This AD Involve a Significant Rule or Regulatory Action?

- I he reasons discussed above, I certify that this AD:
- 1. Is not a "significant regulatory action" under Executive Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- 3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "AD Docket No. 2003-CE-19-AD" in your request.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

F T 39-AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

hority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends Sec. 39.13 by adding a new AD to read as follows:

Regulatory Information

2003-26-14 Kidde Aerospace: Amendment 39-13413; Docket No. 2003-CE-19-AD.

When Does This AD Become Effective?

(a) This AD becomes effective on February 20, 2004.

What Other ADs Are Affected by This Action?

(b) None.

What Airplanes Are Affected by This AD?

- (c) This AD affects aircraft that are certificated in any category and incorporate hand-held halon fire extinguishers with the following:
- (., l'art number (P/N) 898052; and
- (2) A serial number in the range of V-432001 through W-389653 inclusive that were manufactured sometime from 1995-2002.
- (i) Serial numbers are identified by the Underwriter's Laboratories (UL) number printed on the label and are listed in succession.
- (ii) Other variants of the UL number with prefixes other than "V" or "W" are not affected by this AD.

What Is the Unsafe Condition Presented in This AD?

(d) This AD is the result of information that shows that the discharge time of the affected fire extinguishers exceeds the maximum allowable discharge time. The problem is due to incomplete crimping of the siphon tube. We are issuing this AD to remove from service fire extinguishers that have this incomplete crimping of the siphon tube. If not removed from service, these fire extinguishers could function at diminished levels and compromise the level of safety in an emergency situation.

What Must I Do To Address This Problem?

(e) To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
Remove from service any P/N	III	Kidde Aerospace Service Bulletin
898052 hand-held halon fire	February 20, 2004 (the effective date	898052-26-449, dated October 7,

extinguisher that has a serial number of V-432001 through W-389653 usive and was manufactured sometime from 1995-2002. You may not operate any aircraft without the applicable fire extinguishing equipment per FAA regulation. (i) Serial numbers are identified by the Underwriter's Laboratories (UL) number printed on the label and are listed in succession. (ii) Other variants of the UL number with prefixes other than "V" or "W" are not affected by this AD.		2002, specifies procedures for identifying the affected fire extinguishers. Use the procedures in this service bulletin for the returned fire extinguishers. Specifically, do not discharge them or ship them to Kidde Aerospace since a special collection point has already been established. Ensure that you follow all Department of Transportation (DOT) regulations (49 CFR) in the transport of fire extinguishing equipment. The regulations identify fire extinguishers containing compressed or liquefied gas as hazardous.
(2) The owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7) may remove the fire extinguisher specified in paragraph (e) (1) of this AD. Make an entry into the aircraft records showing compliance with this portion of the AD in ordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).		Not Applicable.
(3) Do not install, on any aircraft, a Kidde Aerospace P/N 898052 handheld halon fire extinguisher V-432001 through W-389653 inclusive that was manufactured sometime from 1995-2002.	As of February 20, 2004 (the effective date of this AD).	Not Applicable.

What About Alternative Methods of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.13. Send your request to the Manager, Atlanta Aircraft Certification Office, FAA. For information on any already approved alternative methods of compliance, contact Charles H. Bowser, Flight Test Engineer, FAA, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, Suite 450, Atlanta, Georgia 30349; telephone: (770) 703-6047; facsimile: (770) 703-6097.

Is There Material Incorporated by Reference?

(g) You must do the actions required by this AD per Kidde Aerospace Service Bulletin 898052-26-449, dated October 7, 2002. The Director of the Federal Register approved the incorporation by reference of this service bulletin in dance with 5 U.S.C. 552(a) and 1 CFR part 51. You may get a copy from Kidde Aerospace, Kidde Technologies, Inc., 4200 Airport Drive, NW, Wilson, North Carolina 27896; telephone: (252) 237-7004. You may review copies at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the

Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

Footer Information

Issued in Kansas City, Missouri, December 23, 2003. Michael Gallagher, Manager, Small Airplane Directorate, Aircraft Certification Service. [FR Doc. 04-44 Filed 1-6-04; 8:45 am] BILLING CODE 4910-13-P

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SCHEDULED MAINTENANCE CHECKS - MAINTENANCE PRACTICES

				Check
1.	General		H	CARCK

- A. Prior to commencing inspection, disengage circuit breakers as required to prevent damage to aircraft systems or maintenance personnel.
- B. Record all flight crew complaints and inspection discrepancies on proper paper work.
- C. For Engine Periodic Inspection Requirements, reference Inspection / Check section of applicable AlliedSignal Aerospace Light Maintenance.
- D. Remove inspection plates, fairings and covers as required.
- E. Task may be omitted as scheduled maintenance requirement if an equivalent check/test is performed as part of the operators flight or ground crew preflight procedures.
- F. Before starting inspections, ensure that protective covering is installed over rugs and seat cushions and that all personnel entering aircraft remove shoes or wear wing socks.

Aircraft Owner	
Serial No. <u>368</u>	Registration No. <u>N8356</u> LOANEL
	Right Engine Serial No. P-7736/
Aircraft Hours 8704.2	Landings 6277
Left Engine Hours 8300,5	Cycles
Right Engine Hours <u>4694.</u> 2	Cycles 5 418
Date Inspection Started 1-16.05	Completed / - 25.05



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EFFECTIVITY: ALL

5-20-00 Page 202 Jan 31/2003



NOSE TO FORWARD BULKHEAD - MAINTENANCE PRACTICES

				MECH	INSP	200	400	800
Nose	e to F	orward	d Bulkhead					
A.	Jack	Aircra	aft. Refer to 7-10-00, Maintenance Practices.					
	NOT	<u>Е:</u> Ве	efore jacking aircraft, deactivate the following:					
	(1)	Stan	dby Attitude Gyro.					
			r Aircraft 152, 154, 181, 187 - 244, 246 and 247, Il out Emergency Power Supply Unit.		(D)			
		Att	r Aircraft 245 and 248 - 442, disengage Standby itude Gyro Circuit Breaker, located on Forward lay Box under Copilot Seat.	74	QC-4) ×		
	(2)	diser	and Static Heat - For Aircraft 295 - 442, ngage LH and RH PITOT STATIC HEAT Circuit likers located on Overhead Panel.	74 (RII QC-4) _x		
	(3)	Aircr	e of Attack (AOA) Heat - For model 1124A raft only, <u>295 - 440,</u> disengage AOA Circuit liker located on Overhead Panel.	TH (RII QC-4	x		
B.	Nos	e Land	ding Gear.					
			2-10-04, Maintenance Practices, Tire Servicing e-00, Removal / Installation.		(E.			
	(1)		nove Nose Gear Wheels - match mark wheel and assembly left to right before removal.	(D)	The Co	×		
	(2)		ck Tires - wear, weather checking, oil saturation, flat spots, proper inflation, etc.	92	(A)	×		
	(3)	Check Wheels - corrosion and damage.		90	1,00	X		
	(4)	Whe	eel Axle.	/	(20	0		
		(a)	Check Axle for corrosion (internal and external), damage and evidence of irregular wear.	Pi)(3	×		
		(b)	Check Axle for cracks using dye penetrant inspection method.	NA			x	

EFFECTIVITY: ALL

1.

5-20-01 Page 201 Jan 31/2003



- C. Wheel Bearings, Bearing Cups and Spacers.
 - (1) Remove Bearings clean and inspect.
 - (2) Check Bearing, Cups and Spacers for galling.
 - (3) Pack Bearings. Refer to 12-20-00, Maintenance Practices.
 - (4) Install Nose Wheels match marks aligned.
- D. Check following items for general condition, tube integrity, cracks, corrosion, damage, chafing, security of attachment and leakage:
 - (1) Check Strut for proper service and correct inflation. Refer to 12-10-04, Maintenance Practices.
 - (2) Outer Strut Body. Perform Penetrant Inspection of Actuating Cylinder Attach Lug Root and Upper Bearing Retaining Nut Lock Screw Thread Area. Refer to 32-20-01, Inspection / Check.
 - (3) Scissors and Bushings wear (0.020 inch maximum clearance between bushing at knee-joint).
 - (4) Drag Brace Upper and Lower Lugs and Fittings.
 - (5) Bungee Cables.
 - (a) Condition.
 - (b) Dragbrace Downlock Tension Test. Refer to 32-20-02, Inspection / Check.
 - (6) Retract Cylinder and Attach Points.
 - (7) Inspect Steering Bracket Assemblies for cracks. Refer to 32-50-08, Inspection / Check.
 - (a) Upper Bracket Assembly (ES12970-7 or 2236.0200.000) and Lower Bracket Assembly (ES12970-6 or 2236.0300.000).

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- (b) Upper Bracket Assembly (A01-ES12970-7 or A01-2236.0200.000) and Lower Bracket Assembly (A01-ES12970-6 or A01-2236.0300.000).
- (8) Trunnion Fittings.
- Nose Steering Cylinders and Attach Points.
- (10) Nose Gear Centering Spring and Attach Points.

NOTE: With scissors connected, rotate strut left and right and check for smooth centering movement.

- (11) Nose Steering Linkage Universal Joints.
- (12) Nose Steering Control Valve.
- (13) Nose Steering Control Cables and Pulleys.
 - (a) While rotating NLG from stop to stop, check cables for wear, fraying, strand breakage and security, especially at hidden pulley areas.
 - (b) Check Cable Tension. Minimum tension -19pounds Refer to 32-50-00, Adjustment / Test.
 - (c) Pulleys. Inspect for general condition, cleanliness and free movement.
 - (d) Cable Inspection. Refer to 32-50-07, Inspection / Check.
 - (e) Lubricate Cables. Refer to 12-20-00, Maintenance Practices.
- (14) Gear Uplock Assembly.
- (15) Gear Selector Valve.
- (16) All Hydraulic Lines.
- (17) Electrical Bundles, Microswitches, Wiring and Connections.
- (18) Structure.
- (19) Door Actuating Rods and Rod Ends.

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<u>FUSELAGE - FORWARD PRESSURE BULKHEAD TO AFT PRESSURE BULKHEAD - MAINTENANCE PRACTICES</u>

1.	Fuselage - Forward Pressure Bulkhead to Aft Pressure
	Bulkhead

- A. Instrument Panel Check general condition:
 - (1) Accessible Instruments, Hoses and Lines.
 - (2) Accessible Electrical Wire Bundles and Connections
- B. Pressurization System Check:
 - (1) Accessible Hoses and Lines condition.
 - (2) Isobaric Valve safety wired in open position.
 - (3) Clean Cabin Air Pressure Controller Air Filter. Refer to 21-30-01, Cleaning / Painting.
 - (4) Replace Air Filter Element Cartridges.
- C. Oxygen System Check:
 - (1) Crew Oxygen Masks.
 - For aircraft with EROS Mask and Regulator Assembly, refer to 35-00-00, Adjustment / Test, Paragraph 2A and 2B.
 - For aircraft with Puritan Bennett Mask and Regulator, ref. 35-00-00, Adjustment / Test, Paragraph 2A.
 - (2) Passenger Oxygen Supply System Test Refer to 35-00-00, Adjustment / Test.
- Mindshield, Windows and Pilot Openable Window.
 - (1) Delamination, scratches, cracks, seal and latch.

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MECH INSP 200 400 800 (2) Renew or reactivate desiccant crystals as necessary. If moisture is noted between window panes, perform procedure to remove residual moisture. Refer to 30-40-00, Maintenance Practices, Service Desiccant X System. CAUTION: DO NOT EXCEED 2.0 PSI. TH X (3) Terminal Contact Assemblies for arcing. (4) Check Pilot and Copilot Side Windows for cracks in flange, radius of flange and around all attachment fasteners. Refer to 56-10-03 and 56-10-04, Inspection / Check. X NA NOTE: Not applicable to Cockpit Side Window (343017-507) (pilot) or (343003-501) (copilot), post Service Bulletin 1124-56-113. Windshield Wiper Assembly - condition and security. Refer to 30-40-00, Maintenance Practices. Control Pedestal - general condition. X Indicators. (1) X (2) Controls. Switches. (3) (4) Electrical connections. X Throttles and Piggy Back Levers - ease of operation, Overhead Electrical Panels - general condition. Electrical bundles, connections, damage and X tightness. NA **CAUTION: ENSURE ADEQUATE CLEARANCE** BETWEEN ELECTRICAL CONNECTIONS AND FRAME WHEN INSTALLING PANEL. ATC Transponder Test and Inspection. 2 years Refer to FAR 91.413. Altimeter Systems Tests and Inspections, 2 years Refer to FAR 91.411.

E.

F.

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H.



- (20) Power Brake Valve and Parking Brake Linkage.
- (21) Perform Emergency Hydraulic Indicating System Hydraulic Fuse Functional Test. Refer to 29-30-06, Adjustment / Test.
- E. Lubricate Gear Assembly and Linkage. Refer to 12-20-00
 Maintenance Practices.
- F. Upper and Lower Outer Strut Body Bearings remove, clean, inspect and lubricate. Refer to 32-20-01, Removal / Installabon.
- G. Check the following items in and around nose compartment above nose wheel well for general condition, cracks, corrosion, damage, chafing, security of attachment and leakage:
 - (1) Pitot Tubes and Lines.
 - (2) Drain Pitot / Static Drain Valves and Traps. Refer to 34-10-01, Maintenance Practices.
 - (3) Electrical Components, Wire Bundles, Windshield Heat Resistors and Terminal Strips.
 - (4) All Structure
 - (5) Oxygen Lines and Bottle. Refer to 5-10-00, Maintenance Practices.
 - (6) Avionics Components and Shock Mounts.
 - (7) AC Inverters (forward or aft installation) and Cooling Fan.
 - (8) For aircraft equipped with Collins W-XR-300 Weather Radar - check Crystal Desiccant Bottle. Refer to 34-40-04, Adjustment / Test, Paragraph F, Desiccant Check.
- H. Radome
 - (1) Condition and security.
 - (2) Static Discharge Diverter Strip Bonding. Refer to 23-60-00, Inspection / Check.

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		MECH	INSP	200	400	800
(7)	Door step - condition, operation and presence of stoppers.	TA	De A	×		
Pass	enger compartment - general condition and security.	74	OC.) x		
(1)	Emergency light - check operation, security, cleanliness and connections. Check battery charge. Refer to 12-10-06, Maintenance Practices, Paragraph 3.					
	NOTE: Every 200 hours or not to exceed 3 months or whenever the emergency lights have been operated from the emergency battery for more than 1 hour.	80	RII QC-4	×		
(2)	Refreshment bar, ice chest, galley coat closet, cabinets, tables, etc. for ease of operation and locking.	TH(RII	×		
(3)	Seats and Seat Belts.	TH	200	X		
(4)	Oxygen, Reading Light, and Ventilating Air Console.	TH	Des	X		
(5)	Windows - delamination, scratches and cracks.	R	Q COLL	X		
(6)	Interior lights.	14	965	X		
(7)	Emergency Exit.	W	QC.)x		
	(a) Check Release Mechanism. (Pull release handle, but not necessary to remove exit from aircraft.)	TH	RII QC-4	x		
	(b) Remove Exit - inspect seal and check operation of game tables for clearance.	NA	DII			x
(8)	Certificates.	7/	9C01	X		
(9)	Avionics components - security and connections.	DH.	2000	X		
(10)	Lavatory and Baggage Compartment - condition.	DA.	000	X		
(11)	Lavatory Door- operation.	A	1	X		

(12) Check Portable Fire Extinguisher. Refer to 5-10-00, Maintenance Practices.

P.



_	FP 11	-	
Q.	Flight and	Passenger	Compartment:

- (1) Remove Flight and Cabin Compartment Seats, Divan, Seat Tracks and carpet as necessary to gain access to floor paneling. Thoroughly inspect under floor for corrosion, damage, wear, security, cleanliness and ensure that all under floor drain paths are clear of debris and sealant.
- (2) Flight Control System Pulleys, Brackets, Guards, Bellcranks and Push-pull Rods - condition, operation and security of attachment.
- (3) Control Column Interconnect Cable.
 - (a) Condition and security. Refer to 27-00-00, Inspection / Check.
 - (b) Check Cable Tension. Refer to 27-00-00, Maintenance Practices, Table 2.
- (4) Flight Control Transition Cables between Fuselage Station 153 and Station 269.
 - (a) Condition and security. Refer to 27-00-00, Inspection / Check.
 - (b) Check cable tension. Refer to 27-00-00, Maintenance Practices, Table 2.
- (5) Lubricate Control System. Refer to 12-20-00, Maintenance Practices.
- (6) Lubricate Gustlock Mechanism. Refer to 12-20-00, Maintenance Practices.
- (7) Perform Aileron and Rudder Static Friction Test. Refer to 27-00-00, Maintenance Practices.
- (8) Engine Control Linkage and Teleflex Cable condition, proper routing and security of clamps.
- (9) Plumbing proper routing, chafing, tube integrity, condition and leaks.
- (10) Check general condition and security of Pressurization and Air Conditioning Components under cabin floor.

1ECH	INSP	200	400	800
				×
				х
			/	×
		1		X
	/			
	X/			x
N				x
Ì				x
				x
				×
				x
		1		х
				X

CI INCD 200 400 000

EFFECTIVITY: ALL

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- (11) Inspect Structure under floor below galley and lavatory toilet installation for fluid leakage, corrosion, general condition and cleanliness.
- R. Standby Attitude Gyro Check Standby Attitude Gyro Emergency Power Supply Unit condition every 200 hours or not to exceed 3 months or whenever the standby attitude gyro has been operated from the emergency power supply for more than 45 minutes. Refer to 34-20-09, Adjustment / Test, Paragraph 2A.
- S. Check the following for condition:

Windshield Wipers.

- (1) Oxygen Thermal Discharge Disc.
- (2) Fuselage Skin.
- (3) Cabin Outflow and Safety Valves condition and cleanliness.
- (4) Static Sources.
- (5) Drain Pitot and Static Valves and Traps. Refer to 34-10-01, Maintenance Practices.
- (6) All Drain Holes and Accessible Fittings.
- (7) Antennas condition.
- (8) Accessible Electrical Connections and Components damage, tightness, chafing, fraying and cuts.
- (9) Accessible Electrical Connections and Components damage, tightness, chafing, fraying and cuts.

MECH	INSP	200	400	800
NA				x
83	RII QC.2	×		
TH	RII	×		
TH.	90.00 90.00	X		
H	SA.	×		
TH	205	×		
77/	299	x		
TH	800	×		
JA	QC.A	×		
T#	(QC.	×		

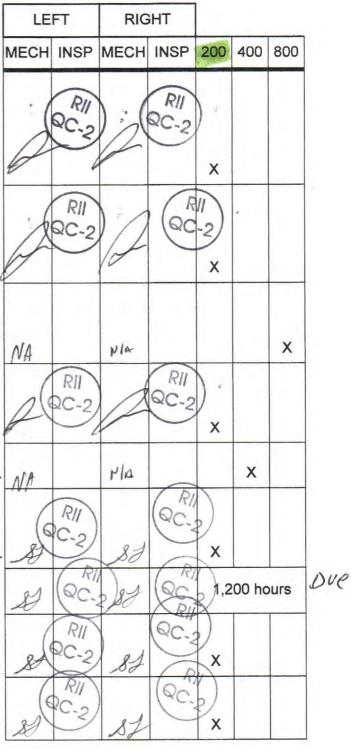
MECH INSP 200 400 800



WING - MAINTENANCE PRACTICES

1. Wing

- Wing Flap general condition, security and cracks.
 - (1) Skin and Rivets.
 - (2) Attach Points:
 - (a) Inspect Hinges and Bearings. Refer to 27-50-00, Inspection / Check, Paragraph 2A.
 - (b) Inspection Hinges and Bearings.
 Refer to 27-50-00, Inspection /
 Check, Paragraph 2B., Steps (2),
 (3) and (4).
 - (c) Bonding Jumpers.
 - Condition, fraying and security.
 - Control Surface Bonding.23-60-00, Inspection/Check.
 - (3) Flexible Drive Shafts.
 - (a) Couple Nuts security, routing, freeplay and structure clearance.
 - (b) Clean, inspect and lubricate. Refer to 27-50-00, Servicing.
 - (4) Actuating Jacks attach points, electrical connections, rigging and microswitch sliders.
 - (5) Flap Position Transmitter
 Potentiometer attach points
 and electrical connection.





(6) Check Flap Vane Segments condition, failed or loose fasteners and security of attach plate. Ref. 27-50-00, Inspection/Check, para. 1.C.

(7)	Check Flap Vane Segments -
, ,	separation of skin from honeycomb.

- (a) Visually inspect upper and lower surfaces for local bulging or looseness of skin.
- (b) Tap test by lightly tapping the skin with a coin or equivalent and comparing the sound at adjacent locations. Separation between skin and honeycomb core can readily be identified by a hollow sound compared to the response at solidly bonded areas. Ref. 27-50-00, Inspection/Check, para. B. (2), (3).
- B. Flap System Time-Exceed Relay Check.
 - (1) Move Flaps to 12° or 20° position.
 - (2) Open Flap Control Circuit Breaker on Overhead Panel. Disconnect Flap Motor Plug P-26.
 - (3) Close Flap Position Indicator and Flap Control Circuit Breakers on Overhead Panel and Flap Circuit Breaker (CB2-5) on No. 2 DC Contactor Box.
 - (4) Activate Flap Selector to DOWN position.
 - (a) On aircraft with a 1A Flap Control CB - Breaker should trip after 20±3 seconds. Repeat procedure in UP position.

LEFT		RIGHT				
MECH		MECH	INSP	200	400	800
	RII QC-2	1	QC.2) x		
J	(RIII)		RII C.2) ×		
NA		NIA			x	
NA		NA			x	
NA		NIA			x	
NA		MA			x	
NA		NA			x	



(b)	On aircraft with a 2A Flap Control
	CB - Breaker should trip after 20
	+ 6 -3 seconds. Repeat
	procedure in UP position.

NOTE: The motor will not run.

- (5) Connect P-26 plug to Flap Motor. Reset Flap Control Circuit Breaker on Overhead Panel, check Flap operation.
- C. Flap Comparator System Test and Adjust. Refer to 27-50-00, Adjustment / Test
- D. Aileron General condition and security.
 - (1) Skin and Rivits.
 - (2) Torque Transfer Tubs.
 - (a) Attach points, safety and exterior for corrosion.
 - (b) Lubricate Rod Endc. Refer to 12-20-00, Maintenance Practices.
 - (3) Aileron Trim Tab to Actuator attach points and electrical connections.
 - (4) Aileron Hinge Points:
 - (a) Check all Bearings looseness, roughness, safety and general condition.
 - (b) Bonding Jumpers
 - Condition, fraying and security.
 - Control Surface Bonding. Refer to 23-60-00, Inspection / Check.

LE	FT	RIG	НТ			
MECH	INSP	MECH	INSP	200	400	800
pla		NIA	4		X	
MA		Ma			×	
NIA	(RII	NA				х
	QC.V	RII		Х		
0	RII QC:V	R)		×		
2	QC.	811		x		
26	RII C-2	A.		x		
S	RII 2C-2	1000	RII	X		
2	RII 2C-2	RII	2)	x		
NA		NA			х	



- Check operation of Fire Protection System. Refer to 26-00-00, Maintenance Practices, Paragraph 1A or 1B.
- J. Battery Temperature System Perform Functional Test of Battery Temperature and Warning System. Refer to 24-30-01, Maintenance Practices.
- K. Check all Internal and External Lights including Cockpit 'Press-to-Test' Lights.
- L. Remove covers from the following listed electrical boxes and check inside for cleanliness, safety and condition of electrical parts.
 - (1) Forward Relay Box (below copilot seat).
 - (2) Fire Control Box (below pilot seat).
- M. Seats, Mountings and Seat Belts Condition.
 - (1) Pilot.
 - (2) Copilot.
- N. Rudder Pedals, Linkage and Bellcrank.
 - (1) Brake Valve Linkages.
 - (2) Freedom of operation.
- O. Cabin Entrance Door. Ref. 52-10-00, Inspection/Check.
 - (1) Ease of operation.
 - (2) Door Hinges.
 - (3) Locking Mechanism.
 - (4) Door Seal condition.
 - (5) Clean and lubricate Door Seals silicone lubricant recommended.

NOTE: Apply only a very light coat of lubricant; excessive lubricant will collect dirt and cause door seal leakage.

(6) Lubricate - Refer to 12-20-00, Maintenance Practices.

MECH	INSP	200	400	800
NA			x	
NIA (QC-4	×		
M	IJ₽.	х		
NA			Х	
NA			X	
NA			x	
NA			X	
NA	ŕ		x	
NA			X	
Or	TH.	x		
No	N	X		
Des	VH	X		
dr	TH	Х		
D	17/	X		
LA	PC PI	x		



- Push-Pull Tube. (5)
 - Inspect external surface of Tube (a) for wear (particularly area of rollers). Maximum reduction of tube crosswise diameter is 0.012". Refer to FAA Airworthiness Directive No. 96-24-11.
 - Clean all Guide Rollers and check Rollers for smooth rotation.
- Lubricate Trim Tab and Servo Tab Hinges from inside with LPS-3 or equivalent.
- (7) Aileron Bellcrank check Travel Stops make contact in both directions of full travel - attach points and safety.
- (8) Check Aileron Control System Trim and Servo Tabs for free-play. Refer 27-10-00, Inspection / Check.
- Static Discharger Wicks and Bases.
 - Condition and security. (a)
 - Static Wick Inspection / Check. 23-60-00, Inspection / Check.
 - Wick Base Bonding Inspection/ Check. Refer 23-60-00, Inspection / Check.
- E. Non-Icing Fuel Vent - obstruction and fuel leakage.
- F. Tip Tank.
 - Tip Tank and Wing Fillet condition, security and fuel leaks.

LEFT		LEFT RIGHT				
MECH	INSP	MECH	INSP	200	400	800
NA		NA			x	
NA		NA			Х	
2	RII 2C-2	2	OC-2	×		
2	OC 22	2	QC-2	×		
NA		NIA				x
2	RII QC-2	2	(QC-2)	×		
NA	/	N/A			X	
N/A		MA			х	
1	50.55 JULIAN	2	QC-	×		
(d	100 P	Jan -	RII QC-	×		



- (2) Vortex Generators security of attachment and / or missing Generators. Refer 57-20-01.
- (3) Navigation Light Lens (Position and Strobe) - condition and security.
- (4) Landing Light condition and security.
- (5) Drain holes cleanliness.
- (6) Static Discharge Wicks and Bases.
 - (a) Condition and security.
 - (b) Static Wick Inspection / Check.23-60-00, Inspection / Check.
 - (c) Wick Base Bonding Inspection / Check. Refer 23-60-00, Inspection / Check.
- (7) Tip Tank Tail Cone Diverter Strip Bonding Inspection / Check. Refer 23-60-00, Inspection / Check.
- F. Wing Skin condition and fuel leakage.
- G. Fuel Tank Drains leakage.
- H. Wing Skin Fairing cracks and loose rivets.
- All Plumbing attached along wing rear spar for proper routing, chafing, tube integrity, condition, security and leaks.

NOTE: Check for clearance between fluid lines and flap flex drive cables and aileron torque tubes while flaps and ailerons are moved through full travel.

 J. Speed Brakes and Lift Dumpers - condition, security, hydraulic leakage and microswitch (R/H outboard only).

		RIG	HT			
MECH	INSP	MECH	INSP	200	400	800
(RII		QC?			
SA	RII	87	3	X		
88	299	3/87	(dt)	3×		
84	OGI	887	SA.	X		
88	600	188	QC O	^		
89	50	-87	SC-5	x		
N/A		NA			х	
NA		N/A			x	
N/A	RII	NA	(5)		x	
87	26.	80	(QE)	×		
80	60%	A	Pil	X		
87		27	600			
826	RII C-2	88	1	×		
_						
NA		A	82	х		



- K. Wire Bundles entering Wing from Aft Fuselage - damage and security.
- L. Check Fuel Dump System:
 - (1) Functional Check Fuel Dump System. Refer 28-00-00, Maintenance Practices, Fuel System Operational Check, Paragraph 4.B.
 - (2) Check Fuel Dump System for operation only (without measuring time/quantity). Refer 28-00-00, Maintenance Practices, Fuel System Operational Check.
- M. Leading Edge and Pneumatic Deicer Boots check condition.

LE	LEFT		RIGHT			
MECH	INSP	МЕСН	INSP	200	400	800
8	(P)	8-5	50.5) 60.51	×		
PIA		NIA		D v 9 4 1,2	98 00 ho	D vurs
NIA		NA				x
S	R11 C-2	0	6C-5	$)_{x}$		



MAIN LANDING GEAR - MAINTENANCE PRACTICES

 Main Landing G 	ear
------------------------------------	-----

NOTE: Steps 1.A. thru 1.D. may be performed at tire change but not to exceed 200 hour intervals.

A. Remove Main Gear Wheels.

Refer 12-10-04, Maintenance Practices, Tire Servicing and 32-41-00, Removal / Installation.

- Tire wear, weather checking, oil saturation, cuts, flat spots, proper inflation, etc.
- (2) Wheel corrosion, damage, overheat indication and wheel half retaining bolt looseness.
- (3) Drive Keys looseness and wear.
- (4) Blow-Out Plugs damage and leakage.
- (5) Wheel Axle.
 - (a) Check Axles for corrosion (internal and external), damage and evidence of irregular wear.
 - (b) Check exposed area for cracks use dye penetrant inspection method.

				Н	RIG	-1	LE
0	800	400	200	INSP	MECH	INSP	ECH
				RII		RII	/
			x	QC-3	ma	OC.	1 MM
-			^			PII	- //
			5 X	(QC	pole	SE SE	1 MA
			X	RII	1414/	C-2)	114
			*	C-2	MUX	RII	114
-			X	(5)	12	OC-	/
MECH		171					M
				Kh	My.	Qh	di,
_			X		- 1		
		.,					1
		X					NA

RIGHT



- (6) Anti-Skid Detectors remove, check for corrosion and contamination. Clean as required and coat axle interior surface with Dow Corning 4 compound (MIL-S-8660B, Amendment 3). Install. Refer 32-44-06, Removal / Installation.
- B. Wheel Bearing, Bearing Cup and Spacer.
 - (1) Remove, clean and inspect Bearing.
 - (2) Inspect Cup for galling and spalling.
 - (3) Pack Bearings. Refer 12-20-00, Maintenance Practices.
- C. Brake Assembly.
 - (1) Brake Mounting Flange cracks.
 - (2) Brake Disc cracks and maximum wear.
 - (3) Brake Housing leakage.
 - (4) Brake Lines chafing, frayed and leakage.
- D. Installation.
 - (1) Install main gear wheels.
 - (2) Check security of drive clip and cap. Check Clip gap is 0.140" ±0.010".
 - (3) Anti-Skid System check.
 - (a) Anti-Skid Control Switch OFF (both Anti-Skid INOP Lights ON)
 - (b) Anti-Skid Control Switch ON (both Anti-Skid INOP Lights OUT)

LE	FT	RIG	НТ			
MECH	INSP	МЕСН	INSP	200	400	800
pule	RII C-2	IMY	QC R	11/3		
		D		Х		
() () () () () () () () () ()	RYL	RII QC-2 RII	Mu	x		
(30	BIL	093	10h	X		
3	am	(ec	1ch	x		
00:5 Sil	Dul	RIJ QC-2	Oh	x		
QC-2	an	QC-2	an	x		
1114	ah	1119	1ah	X		
1114	Qh	IM	Qu-	X		
F. 50.	Ph	RII QC-1	all	x		
60 m	a-	(60	Qu.	X		
P	RII QC-4	B	QC.	×		
PH	RII QC-4	AS	QC-4	×		



- E. Main Landing Gear - check for general condition, cracks, security of attachment and leakage.
 - Strut check for proper service and correct inflation. Refer 12-10-04, Maintenance Practices.
 - (2) Upper Body.
 - (3)Fork.
 - (4) Scissors.
 - Side Brace, Lugs and Fittings.
 - Retract Cylinders and Attaching Points.
 - Jury Brace. (7)
 - Trunnion Retaining Bolts tightness. (8)
 - (9)Microswitches.
 - (10) Electrical Connections and Wire Harnesses.
 - (11) Sealing Compound around Main Wheel Axle Plug Button Grommet and Anti-Skid Conduit upper flared end.
- F. Wheel Well - check for general condition and security.
 - (1) Uplock Assembly leakage.
 - (2) All Fluid Carrying Lines chafing, damage, tube integrity and leakage.
 - Electrical Connections and Wire (3)Bundles.
 - (4) Structure Paint condition and corrosion.
 - (5) Main Gear Doors and Linkage.

LE	FT	RIG	HT			
МЕСН	INSP	MECH	INSP	200	400	800
ī						
NA	6 RH	WA	A	X	X	
JH	C.	24	C.	×		
TH		11/	2000 2000	X		
THE	10 to 1	A		x		
01)_(
NA M	PIP	7#	() A /	X		
7. J.	RAIN	7/	800	X		
TH		TH	200	X		
7/	800	TH	00	×		
TH	1	1#	6	×	-	
DI	(S.C.	A	(oc	X		
14	(R)		60			
	QC.)	Cici			
TH		7#	1	X		
TH	PO 11	TH	000	x		
7/	A	W	(OC.	×		
TH	00	D	2	×		
TH	(OC)	TH	1	X	/	



- (6) Fuel Vent Lines and Clamps.
- (7) Microswitches Uplock and Downlock.
- G. Lubricate Gear Assembly and Linkages. Refer 12-20-00, Maintenance Practices.
 - Remove upper and lower Actuator Attach Bolts and lubricate. Refer 12-20-00, Maintenance Practices.
- H. Landing Gear Functional Tests.
 - (1) Normal System. Refer 32-30-00, Adjustment / Test.
 - (2) Free Fall Extension. Refer 32-30-00, Adjustment/Test, Normal System Functional Test.
 - (3) Emergency Gear Extension Control Cable. Refer 32-30-00, Adjustment / Test.
 - (4) Emergency Extension System. Refer 32-30-00, Adjustment / Test.
- Perform Anti-Skid System Functional Test. Refer 32-44-00, Adjustment / Test.
- Standby Attitude Gyro before lowering aircraft from jacks, activate and check operation of Gyro:
 - For Aircraft 152, 154, 181, 187 244, 246 and 247, install Emergency Power Supply Unit.
 - For Aircraft 245, 248 442, engage Standby Attitude Gyro Circuit Breaker located on the Forward Relay Box under Copilot Seat.

LE	FT	RIG	НТ			
MECH	INSP	МЕСН	INSP	200	400	800
74	3	TH	2	X		
74	PC	TH.		X		
Ph	M	a	P	x		
NA		NA			x	
NA		NA			х	
N/A N/A		NA			x	
PB	PC.4	At	(PC.	×		
NA		NA				X
N/A		NA			X	
27	B			x		



K. Lower Aircraft off jacks. Refer 7-10-00, Maintenance Practices.

NOTE: Ensure all other services and inspection checks required with aircraft on jacks are completed before aircraft jacks are lowered.

NOTE: After lowering aircraft from jacks, reactivate the items listed below:

- (1) Pitot and static heat for Aircraft 295 - 442, engage LH and RH PITOT STATIC HEAT Circuit Breakers located on Overhead Panel.
- (2) Angle of Attack (AOA) Heat For model 1124A Aircraft only, 295 - 442, engage AOA Circuit Breaker located on Overhead Panel.

LE	FT	RIG	НТ			
MECH	INSP	MECH	INSP	200	400	800
87	gr			X		
82	N			x		
8	81			×		



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EFFECTIVITY: ALL

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ENGINE, NACELLE AND PYLON - MAINTENANCE PRACTICES

NOTE: All items in this section (5-20-07) may be inspected at intervals of 300 hours to coincide ith AlliedSignal Engine Inspections at the operators discretion.

Engine, Nacelle and Pylon

- A. Inspect inlet nacelle for cracks, loose nvets, corrosion, security and general condition. Refer 54-00-00, Inspection / Check.
- B. P₂T₂ Sensor.
 - (1) Condition and security.
 - (2) Heater Operational Test. Refer 30-20-00, Maintenance Practices.
- C. Inspect cowl structure, doors, skin and latches for dents, cracks, fit, general condition and operation.
- D. Inspect fire detector element for chafing, kinks, security and general condition.
- E. Inspect low and high pressure bleed duct for leaks, cracks, fit and general condition.

NOTE: Inspect manifold assembly during engine Major Periodic Inspection or whenever the after body is removed.

- F. Inspect the following systems and components for installation, clamping, security, condition, chafing, tube integrity, leakage and safety.
 - (1) Fuel Lines, Fuel Flow Transmitter, Pressure Switch.

LE	FT	RIG	НТ			
MECH	INSP	MECH	INSP	200	400	800
cPB	Pilot W	PS	(\$) (\$) (\$)	×		
Pf		NO		X		
M		B	PO TO	×		
CPB.		PS		×		
Ph		B		X		
NB		CAS		x		
Na		MA				
iPt		PE	9.9	×		



- (2) Hydraulic Lines, Attenuator, Hydraulic Pump and Quick Disconnects.
- (3) Hydraulic Pump Remove Drive Spline, inspect and lubricate. Refer 29-10-00, Inspection / Check.
- (4) Oil Pressure Lines, Pressure Transmitter and Low Pressure Switch.
- (5) Electrical Wining, Connectors.
- G. Inspect Engine Mount and Attachment for security and general condition.
- H. Visually inspect Jet Tail Pipe Nozzles for dents, cracks, bulges and general condition.
- I. Check Engine Throttle System for freedom of movement, Control Cable routing, security of clamps, clearance and general condition.
- J. Check Pylons and Firewalls for cracks, condition of Firewall Sealant, security of Hydraulic, Fuel, Electrical Connections and Mechanical Feed-Throughs.

LE	FT	RIG	SHT			
MECH	INSP	MECH	INSP	200	400	800
LPB(OC PI	P		x		
		ROY-		X		
LPS		B	PII OCA	x		,
UPBQ		13	000	X		
NA		N/A			×	
LPBO		UP (QC,4	×		
NA		NA			×	
N/A		NA			x	



150 HOUR FLIGHT HOURS INSPECTION - MAINTENANCE PRACTICES

NOTE: All items in this section (5-21-00) are to be inspected at 150 hour intervals.

LE	FT	RIG	HT
MECH	INSP	MECH	INSP
LAD	(QC x)	B	Q. 10
LPB		LB CB	(6, %)
LPB 50% REMAY	VENS	LPB 50010 REMAS	\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\
By			

1. Engine

Engine Periodic Inspection. Refer 72-00-00, Inspection / Check section of applicable AlliedSignal Aerospace Light Maintenance Manual.

NOTE: For Engine, Nacelle and Pylon Inspection, Refer 5-20-07, Maintenance Practices.

2. Starter-Generator

- A. Electrical Leads and Cooling Duct condition and security
- B. Cooling Fan nicks and blade damage.
- C. Brushes Wear Check Refer 80-10-00, Maintenance Practices and / or Lucas Aerospace Maintenance Manual, File No. 23700.

3. General

- A. Check that a Preflight Inspection has been made.
- B. Check all applicable FAA Airworthiness Directives and Manufacturers' Service Bulletins. Make required log book entries.
- C. Check that all aircraft documents are current and in order before release of aircraft.

Honeywell

LIGHT MAINTENANCE MANUAL TFE731-2A-3 (REPORT NO. 72-02-15)

TEMPORARY REVISION NO. 72-62

Table 603.1. [All Except 2A, 3-1J]	Routine Periodic Inspection Record For	m
AIRCRAFT SERIAL NO.	DATE	
ENGINE SERIAL NO.	TOTAL ENGINE OPERATING HOURS	
150 HOUR INSPECTION (Refer to Table 603)	MECHANIC LIH RIH	INSPECTOR
Oil and Filter Analysis	VB (195)	
Chip Detector	LPD LP3	70 10 C
[Pre SB 72-3124, Pre SB 72-3128] Accessory Drive Splines – Starter/ Generator and Alternator Drive Splines	NA	
300 HOUR INSPECTION (Refer to Table 603)		/ '/ H
[Pre SB 72-3124, Pre SB 72-3128] Accessory Drive Splines – Starter/ Generator and Alternator Drive Splines	.	
Fan Rotor Assembly Blades		
Fuel Filter Element		
500-800 HOUR INSPECTION		
(Refer to Table 603)		
Electrical Wiring and Connections		,
Plumbing Lines and Connections		
[Pre SB 74-3006] 'Igniter Plugs and Lead Terminals		
[Pre SB 74-3006] Ignition System		
1000 - 1400 HOUR INSPECTION (Refer to Table 603)		
[Post SB 74-3006] Igniter Plugs and Lead Terminals		
[Post SB 74-3006] Ignition System		

Engine Serial: 77482 JET-GARE INTERNATIONAL, INC 3 Saddle Road REPORT DATE: 1/20/2005 Cedar Knolls NJ 07927 STARFLITE MANAGEMENT Telephone: (973) 292-9597 DAVID LENTZ Facsimile: (973) 292-3030 9000 RANDOLPH ST ENGINE MODEL: TFE731-3-1G AIRCRAFT TYPE: WW 1124A HOUSTON, TX 77061 AIRCRAFT SERIAL: CONTACT PHONE: 713-644-1128 368 USA OIL TYPE: MOBIL JET OIL 254 FAX NUMBER: 713-644-8823 MSP # LAB DATE: 2005/1/14 CURRENT NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 1/10/2005 --- FILTER RESULTS ----TSN: 8291 TSO: 0 STST CBST ALST M50 COPR SILV MAGN ALUM GRIT MISC -OIL RESULTS-CYCLES: 5.896 FF AMT: CU ŤR CR AG MG TR AL TR PB ВE MA TAN OIL HOURS: 0.0 0.0 0.0 0.0 0.0 TYPE: CS 0.0 0.0 0.0 0.0 0.00 FILTER HOURS: 155.0 0.0 OIL AND FILTER ANALYSIS FINDINGS ---FORM: FΝ FILTER WEIGHT: B LAB CODE: Q SAMPLE NUMBER: 00718 FLASHPOINT: 0 M. Yazar RECOMMEND: NORMAL LAB COMMENT: SUPPLEMENT TO REPORT JC05A718 SENT 1-17-05 Technician LAB DATE: 2004/6/9 NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 6/7/2004 FILTER RESULTS ----TSN: 8151 TSO: n STST CBST ALST M5D COPR SILV MAGN ALUM GRIT MISC --- OIL RESULTS. CYCLES: 5,795 AMT: ΝI CR AG MG ΤR AL PB TR MA BE TAN OIL HOURS: 140.0 0.0 0.0 TYPE: 0.0 0.0 CS 0.0 0.0 0.0 0.0 0.00 0.0 FILTER HOURS: 140.0 OIL AND FILTER ANALYSIS FINDINGS ----FORM: FN FILTER WEIGHT: 11 MGS LAB CODE: Q SAMPLE NUMBER: FLASHPOINT: 0 RECOMMEND: NORMAL LAB DATE: 2003/8/22 PREVIOUS NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 8/18/2003 ---- FILTER RESULTS -----TSN: 7951 TSO: n -OIL RESULTS. ST ST CB ST ALST M50 COPR SILV MAGN ALUM GRIT MISC CYCLES: 5,640 CU AMT: Νì CR TR AG MG TR AĹ PΒ OIL HOURS: 509.0 TR MΑ 8E TAN 0.0 TYPE: 0.0 0.0 0.0 CS 0.0 0.0 0.0 0.00 0.0 FILTER HOURS: 99.0 AND FILTER ANALYSIS FINDINGS -FORM: FN FILTER WEIGHT: 11 LAB CODE: O SAMPLE NUMBER: 01016 FLASHPOINT: 0 RECOMMEND: NORMAL LAB DATE: 2003/3/28 PREVIOUS NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 3/21/2003 -- FILTER RESULTS ---TSN: 7803 TSO: 0 STST CBST ALST M50 COPR SILV MAGN ALUM GRIT MISC OIL RESULTS CYCLES: 5,535 FΕ CU AMT: CR TR AG MG TR AL TR PB BE MA TAN OIL HOURS: 1,160,0 0.0 0.0 0.0 0.0 TYPE: CS 0.0 0.0 0.0 0,0 0.00 FILTER HOURS: 147.0 AND FILTER ANALYSIS FINDINGS FORM: FN FILTER WEIGHT: 7 LAB CODE: Q SAMPLE NUMBER: 01298 FLASHPOINT: D RECOMMEND: NORMAL

WINSOAP for Honeywell

01/02

/20/2005

Légine Serial: JET-CARE INTERNATIONAL, INC 3 Saddle Road REPORT DATE: 1/20/2005 Cedar Knolls NJ 07927 STARFLITE MANAGEMENT Telephone: (973) 292-9597 **DAVID LENTZ** Facsimile: (973) 292-3030 ENGINE MODEL: TFE731-3-1G 9000 RANDOLPH ST AIRCRAFT TYPE: WW1124 AIRCRAFT SERIAL: 368 HOUSTON, TX 77061 CONTACT PHONE: 713-644-1128 OIL TYPE: MOBIL JET OIL 254 USA FAX NUMBER: 713-644-8823 MSP #: LAB DATE: 2005/1/14 CURRENT NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 1/10/2005 FILTER RESULTS OIL RESULTS TSN: 8641 TSO: 0 CB ST ALST M50 COPR SILV MAGN ALUM GRIT MISC FΕ CU. NI CR CYCLES: 6,170 AG MG PВ 8F TAN AMT: TR TR. MA 0.0 0.0 0.0 0.0 0,0 0.0 0.0 0.0 0.00 OIL HOURS: 0.0 0.0 TYPE: CS OIL AND FILTER ANALYSIS FINDINGS FILTER HOURS: 28.0 FORM: FN FILTER WEIGHT: 21 LAB CODE: Q SAMPLE NUMBER: 00717 M. Yazar FLASHPOINT: 0 Technician RECOMMEND: NORMAL LAB COMMENT: SUPPLEMENT TO REPORT JC05A717 SENT 1/17-05. LAB DATE: 2004/1/30 NORMAL SAMPLE - CONTINUE SENDING SAMPLES AT THE RECOMMENDED INTERVAL SAMPLE DATE: 1/23/2004 -- FILTER RESULTS -OIL RESULTS TSN: 6570 TSO: 0 ST ST CB ST ALST M50 COPR SILV MAGN ALUM GRIT MISC FΕ CU CYCLES: 5,333 MG TAN AMT: TR TR TR MA TR 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.00 OIL HOURS: 32.0 0.0 TYPE: CS SG OIL AND FILTER ANALYSIS FINDINGS FILTER HOURS: 32.0 FORM: FN FΝ FILTER WEIGHT: 13 MGS LAB CODE: Q SAMPLE NUMBER: FLASHPOINT: 0 MAINT PERFORMED: LOANER ENGINE RECOMMEND: NORMAL LAB DATE: 2001/11/19 PREVIOUS UNDETERMINED RECOMMENDATION CODE SAMPLE DATE: 11/13/2001 --- FILTER RESULTS RESULTS TSN: B3 TSO: 0 STIST CB ST ALIST M50 COPR SILV MAGN ALUM GRIT MISC FΕ СП NI CR CYCLES: 0 AG MG AL PB BE TAN AMT: TR TR TR MA 0.2 -0.1-1.0 -1.0 -1.0 -0.1 0.0 OIL HOURS: 83.0 0.0 0.00 0.0 TYPE: CS OIL AND FILTER ANALYSIS FINDINGS FILTER HOURS: LINK FORM: FN FILTER WEIGHT: 15 MGS LAB CODE: Q SAMPLE NUMBER: 00037

PAGE

FLASHPOINT: 0

RECOMMEND: UNKNOWN

MAINT PERFORMED: NO COMMENTS

LAB COMMENT: NO FILTER HOURS WERE SUBMITTED.

	. ,								
		RIZED RE	RELEASE CERTIFICATE 3. Form Tracking N						
		rm 8130-3, AIRWORTHINESS APPROVAL TAG				0000256	158-00001		
4. Organization RAYTHEON AIRCRAFT COMPANY Name and Address 9709 EAST CENTRAL					NY		5. Work Ord Invoice Nu		
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15. Authorized Signature:			16. Approval/ Authorization No:		20. Authorized Signisture : 21 Aq Ce			Approval Centificate No:	
17. Nan	ne (Typed or Printed):		18. Date (m/d/y):		22 Name (Typed or Frinted)				Date (m/d/x/)
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Wor	the user/installer work in actinities. Authority of the country accepts parts/component	untry specified	in block 1 it is es	ssenti	al that the	e user/installer er	sure	that his/her Airv	
	ents in block 14 and 19 do to tion certification issued in a								
FAA	A Form 8130-3 (06-01)	* Insta	ller must cross ch	ieck e	ligibility	with applicable t	echni	cal data.nstaller	inus:

	Approving National Aviation Authority/Country: FAA/United States	FAA Form 8	130-3, AIRWOF	EASE CERTII	FICATI	[+]	3. Form Tracking Num 10105456	ber:
'		NEYWELL INTERNATIONAL - EN O. BOX 52181	GINES, SYSTEMS	& SERVICES (PC413)			5. Work Order/Contrac	t/Invoice Number:
	1	OENIX, AZ 850722181					3128209 002	
	i i	O HARDWARE PRODUCT GROUP						1
	1	RBOR CITY, CA 90710						
6.	tem: 7. Description:	8. Part Number:		9. Eligibility:*	10. Quantity:	11.	Serial/Batch Number:	12. Status/World
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14	Custom It is hereby cert Aviation Administ Production Approv materials are in is on file. Certifies the items identifed above were	ted 08/19/04 by: Angeler P.O. # 4280 ified that (a) the paration approved manufal to manufacture issan airworthy condition manufactured in conformity to: The in a condition for safe operation.	arts and/or acturing ar	materials reflected quality control eywell Internationalship authorization 19. 14 CFR 43.9 Return that the control of the con	systems/meal Inc. and from the I	ethod (b	ods as set fort b) such parts a luction Approva Other regulation specific k 13, the work identified	er Federal th in FAA and/or al Holder fied in Block 13
	Non-approved design data s			Federal Regulations, par return to service.	t 43 and in respe	ct to t	hat work, the items are	approved for
15	Authorized Signature: Hagelina Pa		Authorization No.: 216NM	20. Authorized Signature:		~	21. Approval/Certifi	icate No.:
17	Name (Typed or Printed):	18. Date (m/d/	y):	22. Name (Typed or Printed	d):		23. Date (m/d/y):	
	Angelina Penaloza	08/18/0	4					
		Us	er/Installer Re	sponsibilities				
i	t is important to understand that the exist where the user/installer performs work is Block 1, it is essential that the user/installer Block 1.	n accordance with the national reg er ensures that his/her airworthin	ulations of an airw ess authority accep	vorthiness different than the a ots parts/components/assembli	irworthiness aut ies from the airw	hority orthi	y of the country specific ness authority of the co	untry specified
L	national regulations by the user/installer AA form 8130-3 (6-01)	before the aircraft may be flown.		applicable technical data.	em an mstanado	at cert		
-			on ongionity with	מארוויים ופטוווויםו שמום.			NSN:0052-00	-012-9005

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4	. Organiza	ition Name a			- ENGINES, SYSTEM	S & SERVICES (PC413)		5. Work Order/Contrac	t/Invoice Number
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			1	AZ 850722181 WARE PRODUCT GR	OUD			3128209 002	
		÷		240TH ST.	OUP				:
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			Customer P.	0 # 4200			J		
		It is	hereby certified	that (a) th	ne narts and/or	materials reflecte			
			arb arc in an ar	rworthy cond	dition. Direct	ship authorization	from the I	Production Approva	l Holder
		is on	rile.		•				
14.	Certifies th	ne items ider	ntifed above were manufac	tured in conformity t	to:	19. 14 CFR 43.9 Retu	m to Service	Other regulation specif	it and the District of
	x			_	•			Displace the second specific	ied in Block 13
	X Approved design data and are in a condition for safe operation. Certifies that unless otherwise specified in and described in Block 13 was accomplished.							ed in accordance with 3816 d	JATBlock 12
	Non-approved design data specified in Block 13. Federal Regulations, part 43 and in respec						ct to that work the items are	4, Code of	
						return to service.		, 2.0	approved (of
15.	Authorized	d Signature:	1.	**	proval/Authorization No.:	20. Authorized Signature:		21. Approval/Certifi	cate No.:
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					User/Installer R	esponsibilities			
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ir	a Block 1.		word morning cubus	es arat morner alfw	orniness authority acce	pts parts/components/assembli	es from the airw	orthiness authority of the co	untry specified
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<u>n</u>	ational reg	gulations by	the user/installer before t	he aircraft may be f	lown.			and account account account	-wance with the

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Authori	ing National Aviation ty/Country: J NITED STATES	2.	AUTHO FAA For	RIZED REL m 8130-3, AIRWO	EASE CE	ERTIFICATE PPROVAL TAG		3. Form Tracking Num	ber:
, F	ation Name and Address ONEYWELL INTERNATION 11 S. 34th St, PHOEN	WAL - ENGINES	& SYSTEMS (PC413) -2181)	C/O K 3120 Brea,	APCO E. Enterprise CA 92821	:	5. Work Order/Contrac Number: 910290-00	t/Invoice
6. Item:	7. Description:		. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number	•	12. Status/Work:	
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14. Certifi	es the items identified at	ove were man	ıfactured in conformi	y to:	19. □ 14 CFR	43.9 Return to Service		Other regulation specified i	in Block 13
	pproved design data and		•		and describe	t unless otherwise specified in ed in Block 13 was accomplishulations, part 43 and in respectivice.	n Block 13, th	e work identified in Block	12 f
15. Author	yized Signature:		16. Approval/Aut	thorization No.:	20. Authorized S	Signature:	21. Appro	oval/Certificate No.:	
Na	yl Demai		ODARF602216NM		1:				
17. Name	(Typed or Printed):		18. Date (m/d/y):		22. Name (Type	d or printed):	23. Date ((m/d/y):	
DARYL	GERMAINE	8130dwg	05/03/04						
				User/Installe	r Responsibilit	ies	!	****	
It is impo	tant to understand tha	t the existence	of this document a	lone does not automatics	illy constitute auth	ority to install the part/con	nnonentlesse	ımhlı:	
Where the	user/installer perform	s work in acc	ordance with the na	tional regulations of an	airworthiness auth	ority to install the particol ority different than the airv onents/assemblies from the	vorthings a	ath anima a Callana ann ann a	pecified in ry specified in
Statements the nation	s in Blocks 14 and 19 al regulations by the u	do not constitu ser/installer b	ute installation certi efore the aircraft ma	fication. In all cases, air ay be flown.	craft maintenance	records must contain an ins	stallation cer	tification issued in accor	rdance with

*Installer must cross-check eligibility with applicable technical data

FAA Form 8130-3 (6-01)

FAA-PMA Applied Energy Technology Corp. Tempe, Arizona Mfg. Code 57597

AETC P/N AE13083-5 A lge Life: 10 Years Total

Ci (from month and year on cartridge)

Eligibility for Installation: Aerostar (Smith) PA-60-600 (Aerostar

600), PA-60-601 (Aerostar 601); Avion Marcel Dassault Falcon 10; Raytheon

Corp. Jets, Inc. (formerly Aircraft Corp.) D-18 Series, 35 Series,

C45 Series, 50 Series, 60, A60, B60, 65-80 Series, 90 Series, 95-55 Series, 56TC,

99 Series, 100 Series, 200 Series, 300,

B300, 400, 400A, 1900, 1900C, 1900D; Bell 204B, 205A, 206, 212, 214B, 214ST,

222, 412; Cessna 172 Series, 175 Series, 300 Series, 400 Series 500, 501, 525, 550, 650:

552, S550, 551, 560. Construcciones Aeronauticas, (CASA) CN-235, CN-235-100; Dassault

S.A. Mystere-Falcon 20-C5, -D5, -E5, -F5,

Mystere Falcon 50; de Havilland DHC-6-1, -100, -200, -300; Dornier-Werke GmbH/Dornier Luftfahrt GmbH Do

228-100, 228-101, 228-200, 228-201, 228-202, 228-212; Fairchild SA-26T, SA-26AT, SA226T, SA226-AT, SA226-T(B),

SA227-AT, SA227-TT, SA226-TC, SA227-AC (C-26A), SA227-BC (C-26A),

SAPPC, SA227-CC, SA227-DC Eurocopter Deutschland j); GmbH MBB-BK 117 A-1, A-3, A-4, B-1,

B-2, C-1; Israel Aircraft Industries, Ltd. Astra SPX 1121, 1121A, 1121B, 1123, 1124, 1124A, 1125; Westwind

Astra, Astra SPX, Learjet Learjet 24, 24A, 24B, 24B-A, 24C, 24D, 24D-A, 24E, 24F, 24F-A, 25, 25A, 25B, 25C, 25D, 25F,

28, 29, 35, 35A (C-21A), 36, 36A, 55, 55B, 55C; Piper PA-23, -23-160, -23-235, -23-250, (Navy UO-1), -E23-250,

PA-31, -31-300, -31-325, -31-350, -31P, -31T, -31T1, -31T2, -31T3, -31P-350; Raytheon Corporate Jets DH.125-1A,

1A/ 522, DH.125-1A/ S-522, DH.125-3A, -3A/R, DH.125-1A/R-522, DH.125-3A/RA, DH.125-400A BAe 125-800A, -1000A; Sabreliner NA-265-60, -80.

FAA-PMA
Applied Energy Technology Corp.
Tempe, Arizona
 Mfg. Code 57597
AETC P/N AE13083-5 A
C. dge Life: 10 Years Total
(from month and year on cartridge)
Eligibility for Installation:
Aerostar (Smith) PA-60-600 (Aerostar
600), PA-60-601 (Aerostar 601); Avion
Marcel Dassault Falcon 10; Raytheon
Corp. Jets, Inc. (formerly Beech
Aircraft Corp.) D-18 Series, 35 Series,

C45 Series, 50 Series, 60, A60, B60, 65-80 Series, 90 Series, 95-55 Series, 56TC, 99 Series, 100 Series, 200 Series, 300, B300, 400, 400A, 1900, 1900C, 1900D; Bell 204B, 205A, 206, 212, 214B, 214ST, 222, 412; Cessna 172 Series, 175 Series, 300 Series, 400 Series 500, 501, 525, 550, 552,

(CASA) CN-235, CN-235-100; Dassault Mystere-Falcon 20-C5, -D5, -E5, -F5, Mystere Falcon 50; de Havilland DHC-6-1, -100, -200, -300; Dornier-Werke GmbH/Dornier Luftfahrt GmbH Do 228-100, 228-101, 228-200, 228-201, 228-202, 228-212; Fairchild SA-26T, SA-26AT, SA226T, SA226-AT, SA226-T(B), SA227-AT, SA227-TT, SA226-TC, SA227-AC (C-26A), SA227-BC (C-26A),

SA227-CC,

GmbH MBB-BK 117 A-1, A-3, A-4, B-1, B-2, C-1; Israel Aircraft Industries, Ltd. Astra SPX 1121, 1121A, 1121B, 1123, 1124, 1124A, 1125; Westwind Astra, Astra SPX; Learjet Learjet 24, 24A, 24B, 24B-A, 24C, 24D, 24D-A, 24E, 24F, 24F-A, 25, 25A, 25B, 25C, 25D, 25F, 28, 29, 35, 35A (C-21A), 36, 36A, 55, 55B, 55C; Piper PA-23, -23-160, -23-235, -23-250, (Navy UO-1), -E23-250, PA-31, -31-300, -31-325, -31-350, -31P, -31T, -31T1, -31T2, -31T3, -31P-350; Raytheon Corporate Jets DH.125-1A, 1A/ 522, DH.125-1A/ S-522, DH.125-3A, -3A/R. DH.125-1A/R-522, DH.125-3A/RA, DH.125-400A BAe 125-800A, -1000A; Sabreliner NA-265-60, -80.

Eurocopter Deutschland

S550,

551.

Construcciones

SAZ PC,

560,

Aeronauticas,

650;

S.A.

SA227-DC

FAA-PMA Applied Energy Technology Corp. Tempe, Arizona Mfg. Code 57597 AETC P/N AE13083-5 A Codge Life: 10 Years Total

dge Life: 10 Years Total (from month and year on cartridge)

(from month and year on cartridge Eligibility for Installation:

Aerostar (Smith) PA-60-600 (Aerostar 600), PA-60-601 (Aerostar 601); Avion Marcel Dassault Falcon 10; Raytheon Corp. Jets. Inc. (formerly Beech

Marcel Dassault Falcon 10; Raytheon Corp. Jets, Inc. (formerly Beech Aircraft Corp.) D-18 Series, 35 Series, C45 Series, 50 Series, 60, A60, B60, 65-

Aircraft Corp.) D-18 Senes, 35 Senes, C45 Series, 50 Series, 60, A60, B60, 65-80 Series, 90 Series, 95-55 Series, 56TC, 99 Series, 100 Series, 200 Series, 300, P200, 400, 400A, 1900, 1900C, 1900D-

99 Series, 100 Series, 200 Series, 300, B300, 400, 400A, 1900, 1900C, 1900D; Bell 204B, 205A, 206, 212, 214B, 214ST, 222, 412; Cessna 172 Series, 175 Series,

222, 412; **Cessna** 172 Series, 175 Series, 300 Series, 400 Series 500, 501, 525, 550, 551, 552, 560, 650; **Construcciones** Aeronauticas, S.A.

Construcciones Aeronauticas, S.A. (CASA) CN-235, CN-235-100; Dassault Mystere-Falcon 20-C5, -D5, -E5, -F5, Mystere Falcon 50; de Havilland DHC-

Mystere-Falcon 50; de Havilland DHC-6-1, -100, -200, -300; Dornier-Werke GmbH/Dornier Luftfahrt GmbH Do 228-100, 228-101, 228-200, 228-201, 228-

202, 228-212; **Fairchild** SA-26T, SA-26AT, SA226T, SA226-AT, SA226-T(B), SA227-AT, SA227-TT, SA226-TC, SA227-AC (C-26A), SA227-BC (C-26A).

SA227-AC (C-26A), SA227-BC (C-26A), SA227-PC, SA227-CC, SA227-DC ((;); Eurocopter Deutschland GmbH MBB-BK 117 A-1, A-3, A-4, B-1,

GmbH MBB-BK 117 A-1, A-3, A-4, B-1, B-2, C-1; Israel Aircraft Industries, Ltd. Astra SPX 1121, 1121A, 1121B,

Ltd. Astra SPX 1121, 1121A, 1121B, 1123, 1124, 1124A, 1125; Westwind Astra, Astra SPX; Learjet Learjet 24, 24A, 24B, 24B-A, 24C, 24D, 24D-A, 24E, 24F, 24F-A, 25, 25A, 25B, 25C, 25D, 25F,

24A, 24B, 24B-A, 24C, 24D, 24D-A, 24E, 24F, 24F-A, 25, 25A, 25B, 25C, 25D, 25F, 28, 29, 35, 35A (C-21A), 36, 36A, 55, 55B, 55C; **Piper** PA-23, -23-160, -23-235, -23-250, (Navy UO-1), -E23-250,

55B, 55C; **Piper** PA-23, -23-160, -23-235, -23-250, (Navy UO-1), -E23-250, PA-31, -31-300, -31-325, -31-350, -31P, -31T1, -31T1, -31T2, -31T3, -31P-350; **Raytheon Corporate Jets** DH.125-1A,

1A/ 522, DH.125-1A/ S-522, DH.125-3A, -3A/R, DH.125-1A/R-522, DH.125-3A/RA, DH.125-400A BAe 125-800A, -1000A; Sabreliner NA-265-60, -80.

FAA-PMA Applied Energy Technology Corp. Tempe, Arizona Mfg. Code 57597 **AETC P/N AE13083-5 A** c´ dge Life: 10 Years Total (from month and year on cartridge) Eligibility for Installation: Aerostar (Smith) PA-60-600 (Aerostar 600), PA-60-601 (Aerostar 601); Avion Marcel Dassault Falcon 10; Raytheon Corp. Jets, Inc. (formerly Aircraft Corp.) D-18 Series, 35 Series, C45 Series, 50 Series, 60, A60, B60, 65-80 Series, 90 Series, 95-55 Series, 56TC, 99 Series, 100 Series, 200 Series, 300, B300, 400, 400A, 1900, 1900C, 1900D; Bell 204B, 205A, 206, 212, 214B, 214ST, 222, 412; Cessna 172 Series, 175 Series, 300 Series, 400 Series 500, 501, 525, 550, 552, S550. 551, 560, 650: Construcciones Aeronauticas, (CASA) CN-235, CN-235-100; Dassault Mystere-Falcon 20-C5, -D5, -E5, -F5, Mystere Falcon 50; de Havilland DHC-6-1, -100, -200, -300; Dornier-Werke

GmbH/Dornier Luftfahrt GmbH Do 228-100, 228-101, 228-200, 228-201, 228-202, 228-212; Fairchild SA-26T, SA-26AT, SA226-T, SA226-AT, SA226-TC, SA227-AT, SA227-TT, SA226-TC, SA227-AC (C-26A), SA227-BC (C-26A), SA227-DC (C-26A); SA2

Ltd. Astra SPX 1121, 1121A, 1121B, 1123, 1124, 1124A, 1125; Westwind Astra, Astra SPX; Learjet Learjet 24, 24A, 24B, 24B-A, 24C, 24D, 24D-A, 24E, 24F, 24F-A, 25, 25A, 25B, 25C, 25D, 25F, 28, 29, 35, 35A (C-21A), 36, 36A, 55, 55B, 55C; Piper PA-23, -23-160, -23-235, -23-250, (Navy UO-1), -E23-250, PA-31, -31-300, -31-325, -31-350, -31P, -31T, -31T1, -31T2, -31T3, -31P-350; Raytheon Corporate Jets DH.125-1A, 1A/522, DH.125-1A/S-522, DH.125-3A, -3A/R, DH.125-1A/R-522, DH.125-3A/RA, DH.125-400A BAe 125-800A, 1000A; Sabreliner NA-265-60, -80.

Israel Aircraft Industries,

B-2, C-1;

	ing National Aviation hority/Country:	2.						3. Form T	racking Number:
	VUnited States	AUTH		ZED RELI m 8130-3, AIRWOR			ICATE	Ā	AETC 0502
4. Organi:	zation Name and Address:					THO VAL TAG		5. Work C	Order/Contract/Invoice
				gy Technology C	orp.			Number:	
				ardy Drive #20				AET	C Invoice #3589
:		Ten	ipe, AZ.	USA 85282-1924	,	(PQ1989NM)			
6. Item:	7. Description:	8. Part Nu	mber:	9. Eligibility: *		10. Quantity:	11. Serial/Batch Num	ber: 12.	Status/Work:
1	Cartridge, Fire	e AE130	83-5	Aerostar (Smith) P	A-60-600	289	Lot AEN 1-78	3	NEW
	Extinguisher		i i	(Aerostar 600),					
				(for complete list o see attachment)	f eligibility				
13. Ren	narks:					.l	<u> </u>		
	FAA-PMA PA								
#* ****	AIRWORTH	INESS APPROV	AL-PAR	TS. END. —					
		•							.1
							4		
							;		
14. Certi	fies the items identified abo	ove were manufacture	d in conformi	ity to:	19. ☐ 140	FR 43.9 Return to	Service 🛘 Oth	er regulatio	n specified in Block 13
(A)	A	1 1 1141 4			Certifie	that unless otherv	rise specified in Block 13	the work id	lentified in Block 12
	Approved design data an Non-approved design dat		-	on.	and des	cribed In Block 13 v	was accomplished in acco	rdance with	Title 14, Code of
_			•			o service.	3 and in respect to that	work, the ite	ms are approved for
15. Auth	orized Signature:	_	16. Approv	val/Authorization No.:	20. Authoriz	ed Signature:	$\overline{}$	21. Appr	roval/Certificate No.:
1	A	Shund	DMI	R602521NM		_		↓ "	
·		/mw·	Divi						
17. Name	e (Typed or Printed):		18. Date (n	n/d/y):	22. Name (T	yped or Printed):	•	23. Date	(m/d/y):
	Michael A. Sch	uetz		10/18/04				ļ	
				User/Installer	Responsibili	ties	***************************************		
It is impo	ortant to understand that t	he existence of this doc	ument alone	does not automatically	constitute auth	ority to install the p	art/component/assembly		
Where the Block 1, Block 1.	ne user/installer performs v it is essential that the user/	vork in accordance wi installer ensures that l	th the nations is/her airwoi	al regulations of an airv thiness authority accep	vorthiness authorits parts/compo	ority different than nents/assemblies fro	the airworthiness authorom the airworthiness aut	ity of the co hority of the	untry specified in country specified in
Statemer national	nts in Blocks 14 and 19 do regulations by the user/ins	not constitute installati taller before the aircra	on certificati ift may be flo	on. In all cases, aircraf wn.	t maintenance r	ecords must contai	n an installation certifica	tion issued i	n accordance with the

2105 South Hardy Drive, Suite 20 • Tempe, Arizona 85282-1924 • (480) 894-1719 • FAX (480) 894-8375

SALES	ORDER NUMBER	3.	361		
CUSTO	OMER P.O. NO	8:	28		
LOTN	IO	AEN 1-7	8		
LOAD	DATE	10/04			• .

		CEF	RTIFICATIO	<u>ON</u>	
TO:	Flight Safety Equipme P.O. Box 4468 115 Village Place Ste. Dillon, CO 80435-446	A			
SELLI	ER CERTIFIES THAT	:			
1)	These materials or Government and Buy purchase order.	parts wer ver Specific	re produced in cations as refere	n conformance venced in, or furnis	vith all contractual shed with, the above
2)	The material or parts produced either from chemical and/ or phy specifications.	materials	s for which the	seller has availa	pie for examination
3)	All processes required performed by a facil Government Agency specification.	lity enecifi	ically approved	or certified by t	ne seners cognizant
I	PART NUMBER		QUANTITY	DI	ESCRIPTION
_AE	TC P/N AE13083-5 A	<u>-</u>	- 289		Cartridge
Ch	m A. /	m	, 	October	
Q.C	AUTHORIZED SIG Mike Schuetz	NATURE		DA	TE
EODY(#	· 171		REVISED 8/9	99	

26 YEARS OF EXCELLENCE Pyrotechnic and Explosive Components 2004

FORM #171

~	•					
Aut	oving National Aviation hority/Country: UNITED STATES A	AUTHORIZED RELEA FAA Form 8130-3, AIRWORTH			3. Form Tracking Nur FD2MA03	
4. Orgai	nization Name and Address: Duncan Aviation	n/Lincoln Airport/Lincoln, NE 6	8524	5. Wo Numb	ork Order/Contract/Invoice Der: FD2MA	
6	7. Description:	T T T T T T T T T T T T T T T T T T T	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
325	AUTOPILOT PANEL	622-0141-023	N/A	001	2539	Inspected
Dis Prese: Co: ma:	rviceable conditions intenance manual CHNICIAN: MATTHEW	ion test. gs: Bench checked ur ion. : Function tested ur	nit per manufac	work o	order.	
14. C		ere manufactured in conformity to: are in a condition for safe operation.	19. X 14 CFR 43.9 Retu Certifies that unless othe in Block 12 and describ with Title 14. Code of F	rwise spec	cified in Block 13, the	ed in accordance

Non-approved design data specified in Block 13. that work, the items are approved for return to service. Duncan Ql 43 21. Approval/Certificate No.: 16. ApprovalAuthorization No.: 20. Authorized Signature: 15. Authorized Signature: JGVR194F 23. Date (m/d/y): 9/21/04 22. Name (Typed or Printed): 18. Date (m/d/y): 17. Name (Typed or Printed): STEVE J. KRINGS User/Installer Responsibilites

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country d in Block 1. sį.

Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving National Aviation 3. Form Tracking Number: -Authority/Country: **AUTHORIZED RELEASE CERTIFICATE** FEAXA0325001 FAA/UNITED STATES FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG 5. Work Order/Contract/Invoice 4. Organization Name and Address: **FEAXA** Duncan Aviation/Lincoln Airport/Lincoln, NE 68524 10. Quantity: 11. Serial/Batch Number: 12. Status/Work: Eligibility: * Part Number: 622-3108-016 N/A001 2481 Inspected 325 AUTOPILOT AMPLIFIER 13. Remarks: CUSTOMER: DUNCAN AVIATION Discrepancy: Recertify. Preliminary Findings: Bench tested and found the unit serviceable. Corrective Actions: Unit function tested per Manufacturer's Maintenance Manual. TECHNICIAN: MATTHEW T. BAKER This document constitutes a signed copy of the work order.

14. Certifies the items identified above	were manufactured in conformity to:	19. X 14 CFR 43.9 Return to Service Other regulation specified in Block 1				
	nd are in a condition for safe operation. ata specified in Block 13.	in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.				
15. Authorized Signature:	16. ApprovalAuthorization No.:	20. Authorized Signature: Q1 43	21. Approval/Certificate No.: JGVR194F			
17. Name (Typed or Printed): 18. Date (m/d/y):		22. Name (Typed or Printed): STEVE J. KRINGS	23. Date (m/d/y): 1/11/05			
	User/Insta	ller Responsibilites				

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

he user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

AUTOPILOT COMPUTER 522-2901-016 N/A 001 2059 Inspected COMPUTER	, .	- 1	2.				3. Fo	orm Tracking Nur	mber:	
Duncan Aviation/Lincoln Airport/Lincoln, NE 68524 7. Description: 8. Far Number: 9. Eligibility * 10. Quantity. 11. Serial/Buck Number: 12. Sansa/Work: 13. Remarks: CUSTOMER: DUNCAN AVIATION Discrepancy: N/C exchange. Bench checked unit and found in serviceable condition. Corrective Actions: Function tested unit per Collins AFC-80() Autopilot Computer Instruction Book. TECHNICIAN: ANDREW BERG This document constitutes a signed copy of the work order. 14. Certifies the items identified above were manufactured in conformity to: Approved design data and are in a condition for safe operation. Non-approved design data specified in Block 13. 15. Authorized Signature: 16. Approval/Authorization No.: 20. Authorized Signature: 21. Approval/Certificate No.: JGVRIPS: 22. Name (Towed or Printed): 23. Date (mid/dy): 24. Name (Towed or Printed): 24. Name (Towed or Printed): 25. Date (mid/dy): 26. Name (Towed or Printed): 25. Date (mid/dy): 26. Name (Towed or Printed): 27. Name (Towed or Printed): 28. Name (Towed or Printed): 27. Name (Towed or Printed): 28. Name (Towed or Printed):	•	AUTHORIZED RELEASE CERTIFICATE NV2840001001								
A. Certifies the tiems identified above were manufactured in conformity to: Approved design data and are in a condition of safe operation. A. Certifies the tiems identified above were manufactured in conformity to: Approved design data specified in Block 13. Approved design data specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved design data specified in Block 13. A. Certifies Specified in Block 13. Approved de	l. Organ		Aviation/L	incoln Airport/Lincoln, NE	68524					
AUTOPILOT 522-2901-016 N/A 001 2059 Inspected COMPUTER 3. Remarks: CUSTOMER: DUNCAN AVIATION Discrepancy: N/C exchange. Bench check. Preliminary Findings: Bench checked unit and found in serviceable condition. Corrective Actions: Function tested unit per Collins APC-80() Autopilot Computer Instruction Book. TECHNICIAN: ANDREW BERG This document constitutes a signed copy of the work order. 4. Certifies the items identified above were manufactured in conformity to: Approved design data and are in a condition for safe operation Non-approved design data specified in Block 13. Approved design data specified in Block 13. Certifies the items identified above were manufactured in conformity to: Approved design data and are in a condition for safe operation Non-approved design data specified in Block 13. Certifies the items identified above were manufactured in conformity to: Approved design data and are in a condition for safe operation Non-approved design data specified in Block 13. Certifies the items identified above were manufactured in conformity to: Certifies that unless otherwise specified in Block 13 was accomplished in accordance with Tuel 4. Code of Federal Regulations, 43 and in respect to that work, the items are approved for return to service. Certifies the items identified above were manufactured in conformity to: Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Tuel 4. Code of Federal Regulations, 43 and in respect to that work, the items are approved for return to service. Certifies the items identified above were manufactured in conformity to: Certifies that unless otherwise specified in Block 13, the work identified in Block 13 and accordance with Tuel 4. Code of Federal Regulations, 43 and in respect to that work, the items are approved for return to service. Certifies the items identified above were manufactured in	- 5. (1				10. Quant	ity: 11. Seria	l/Batch Number:	12. Status/Work:	
Discrepancy: N/C exchange. Bench check. Preliminary Findings: Bench checked unit and found in serviceable condition. Corrective Actions: Function tested unit per Collins APC-80() Autopilot Computer Instruction Book. TECHNICIAN: ANDREW BERG This document constitutes a signed copy of the work order. 14. Certifies the items identified above were manufactured in conformity to: Approved design data and are in a condition for safe operation. Non-approved design data specified in Block 13. Non-approved design data specified in Block 13. 15. Authorized Signature: 16. Approval/Authorization No.: 20. Authorized Signature: 21. Name (Typed or Printed): 22. Name (Typed or Printed): 23. Date (mid/9):	_ ^	î .	!	522-2901-016	N/A	002	1 2059)	Inspected	
Approved design data and are in a condition for safe operation. Non-approved design data specified in Block 13. Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service. 15. Authorized Signature: 16. ApprovalAuthorization No.: 20. Authorized Signature: 21. Approval/Certificate No.: JGVR194 F 22. Name (Typed or Printed): 23. Date (m/d/y):	TECHNICIAN: ANDREW BERG									
17. Name (Typed or Printed): AARON J. SPULAK 12/28/04/04 12/28/04 12/28/04 12/28/04 12/28/04 12/28/04 12	in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service. 15. Authorized Signature: 20. Authorized Signature: 21. Approval/Certificate No.: JGVR194F									
Liser/Installer Responsibilites	1/. Na	ame (1 yped or Printed):			AARON J. SPUI			23. 240 (12/28/04	

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country in Block 1.

Statements in blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.



Texas Aircraft Instruments, Inc.

12101 Blume Avenue, Bldg. 396 Houston, Texas 77034 (281) 484-TEXAS (8392) (281) 484-0595 Fax FAA Repair Sta. #HE2R869K

WORKORDER

DATE	W/O NO.
1/17/2005	23690

BILL TO	n-positi	SHIP TO
STARFLITE 9000 RANDOLPH HOUSTON, TX 77061 713-644-1128	(no.	

1 ALTIMETER, KOLLSMAN 1 Sold Heart flood 2 ENCODING ALTIMETER, COLLINS 1 Lested + Certifical 1 Lested + Certifical 1 Lested + Certifical									
1 ALTIMETER, KOLLSMAN 1 Jested Hertsleel 2 ENCODING ALTIMETER, COLLINS 1 Jested Hertsleel 3 G22-3975-003 1152		P.O. NO.	TERMS	DUE DATE	SHIP DATE	SHIP VIA	SH TO THE SHAPE	FOB	
DESCRIPTION PART NUMBER SERIAL NUMBER MECH IN ALTIMETER, KOLLSMAN B4420210014 501 Tested + Certifical Lested + Certifical Tested + Certifical			Net 30	2/16/2005	1/17/2005	200		12	
I sted Hertsheel 2 ENCODING ALTIMETER, COLLINS 1152 Tested + Certified Tested + Certified	ITEM					R SERIAL NU	JMBER	MECH	INSP
Tested + Certified	1	ALTIMETER	R, KOLLSMAN		B4420210	014 5	501	M	外上
2 AIR DATA COMPLITER COLVINS 622-5465-214 1080 MIL	2	ENCODING	ALTIMETER, COL	Lins	622-3975-	003 1	152	G	UJ)
Tax	3		COMPUTER, COLY	INS	622-5465-	214 1	080	M	70.00

ALTIMETER'S AND AIR DATA COMPUTOR CERTIFIED TO COMPLY WITH FAR PART 43 APPENDIX E TO 50,000 FT.

INSPECTOR M.R. ISSUED: \$0.00

	JOSCARD FOR WORKORDERS	23690	
NAME	STARFLITE	DATE PO#	1-19-05
POUNCOS		TELEPHONE	
		POC	
tem#	Description	Part Number	Serial Number
1	ALTIMETER; KOLLSMAN	B4420210014	501
Fittings	YES	Date Promised	ASAP
omplaint Preliminary	TEST		
Inspection	ОК		
Hidden Damage	NONE		
Work Done	TESTED AND CERTIFIED	·	
Parts Used	NONE		
MAINTAINAN	NCE RELEASE) TESTED & CERTIFIED REPAIRED, TESTED	& CERTIFIED OVER	HAULED & CERTIFIEI

JOBCARD FOR WORKORDER#

TAI FORM # 2

LEXAS AIRCKAF I INSTRUMENTS, INC. 12101 BLUME AVENUE HOUSTON, TEXAS 77034 ~ (281) 484-TEXAS FAA CRS# HE2R869K

·	pproving National Aviation Authority/Country: FAA/United States	2. AUTHORIZI FAA Form 813						23690 - 1 orm Tracking Number:
4. Organi		TEXAS AIRCRAFT INSTRUME 12101 BLUME AVENUE, HOUS						ork Order/Contract/Invoice aber: 23690
6. Item:	7. Description	8. Part Number:	9. Eligib	ility*	10. Quantity	11. Serial/B Number:	atch	12. Status/Work:
110	ALTIMETER; KOLLSMAN	B4420210014	VARI	ous	1	501		TESTED
13. Rema	rks:		1		<u> </u>			
Cl	ies the items identified above	Y WITH FAR, PART of the were manufactured in confining a condition for safe operation.	ormity to:	19. Certifies	i4 CFR 43.9 Ret that unless other	um to Service rwise specified in	Block 13,	gulation specified in Block 13 the work identified in Block
N	on-approved design data spe	cified in Block 13.		of Feder				cordance with Title 14, Code work, the items are approved
15. Author	ized Signature:	16. Approval/Authorization No	.:		ized Signature:			ral/Certificate No.: HE2R869K
17. Name(*	Typed or Printed):	18. Date(m/d/y):		22. Name	Phong	1):	3. Date(m	- 05
		User/In	staller Re	esponsib	ilities			
Where the specified in	user/installer performs work in a	ice of this document alone does in occordance with the national regu- iser/installer ensures that his/her	lations of an	airworthine	ss authority diffe	erent than the airv	orthiness :	authority of the country

FAA Form 8130-3(6-01)

accordance with the national regulations by the user/installer before the aircraft may be flown. *Installer must cross-check elegibility with applicable technical data.

Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in

NSN: 0052-00-012-9005

	JOBCARD FOR WORKDRDERS	23690	
NAME	STARFLITE	DATE	1-18-05
ADDRESS		PO#	
		TELEPHONE	
		POC	
tem#	Description	Part Number	Serial Number
2	ENCODING ALTIMETER ; COLLINS	622-3975-003	1152
-ittings	NONE	Date Promised	AS AP
omplaint	TEST		•
Preliminary Inspection	OK		
Hidden Damage	NONE		
Work Done	TESTED AND CERTIFIED	·	
Parts Used	NONE		
4 - 10 - 10 - 10 - 10 - 10 - 10 - 10 - 1	JOS DEI SADE		
AAINTAINAI	NCE RELEASE) TESTED & CERTIFIED (/) REPAIRED, TESTEI	D& CERTIFIED OVER	HAULED & CERTIFIED
Mech	× / Y//////////////////////////////////	nspector	

JOBCARD FOR WORKORDER

TAI FORM # 2

TEXAS AIRCRAFT INSTRUMENTS, INC. 12101 BLUME AVENUE HOUSTON, TEXAS 77034 ~ (281) 484-TEXAS FAA CRS# HE2R869K

,	pproving National Aviation Authority/Country: FAA/United States	2. AUTHORIZ FAA Form 813					23690 - 2 3. Form Tracking Number:
	zation Name and Address:	TEXAS AIRCRAFT INSTRUME 12101 BLUME AVENUE, HOU					5. Work Order/Contract/Invoice Number: 23690
6. Item:	7. Description	8. Part Number:	9. Eligi	bility*	10. Quantity	11. Serial/Batc Number:	h 12. Status/Work:
2	ENCODING ALTIMETER	622-3975-003	VAR	MOUS	1	1152	TESTED
14. Certif	ies the items identified abo	ove were manufactured in cone in a condition for safe operate cified in Block 13.	formity to:	19. Certifies	14 CFR 43.9 Restant unless other	turn to Service O rwise specified in Blo k 13 was accomplish	ther regulation specified in Block 13, the work identified in Block ed in accordance with Title 14, Code to that work, the items are approved
15. Author	ized Signature:	16. Approval/Authorization N	o.:	for retur	n to service. rized Signature:		Approval/Certificate No.: HE2R869K
17. Name(Typed or Printed):	18. Date(m/d/y):		22. Name	Typed or Printer		Date(m/d/y): - 19-05
		User/I	nstaller I	Responsit	oilities		
Where the specified is the country Statements	user/installer performs work in n Block 1, it is essential that the y specified in Block 1. in Blocks 14 and 19 do not co	tence of this document alone does a accordance with the national reg e user/installer ensures that his/he astitute installation certification.	not automat gulations of a r airworthing In all cases,	ically constitute in airworthing in airworthing in authority in aircraft main	tute authority to ess authority diff accepts parts/co	rement than the airwort imponents/assemblies	from the airworthiness authority of
accordance	e with the national regulations last 130-3(6-01)	by the user/installer before the air *Installer must cross-check ele	craft may be	flown.			52-00-012-9005

	JOBCAND FOR WORKONDERS	23690	
NAME	STARFLITE	DATE	1-18-05
ADDRESS		PO#	
		TELEPHONE	
		POC	
item#	Description	Part Number	Serial Number
3	AIR DATA COMPUTOR; COLLINS	622-5465-214	1080
Fittings	YES 2 BLUE	Date Promised	ASAP
Complaint	TEST		
Preliminary inspection	ок		
Hidden Damage	NONE		
Work Done	TESTED AND CERTIFIED	•	
Parts Used	NONE		
MAINTAINA	NCE RELEASE		
	TESTED & CERTIFIED REPAIRED, TESTE	D & CERTIFIED OVER	HAULED & CERTIFIED
Meci	manic / 7/1/4/1/01/1/01/	Inspector W	

JOBCARD FOR WORKORDER#

TAI FORM # 2

TEXAS AIRCRAFT INSTRUMENTS, INC. 12101 BLUME AVENUE HOUSTON, TEXAS 77034 ~ (281) 484-TEXAS FAA CRS# HE2R869K

	Authority/Country: FAA/United States	FAA Form 813					3. I	23690 - 3 Form Tracking Number:
4. Organ		TEXAS AIRCRAFT INSTRUME 12101 BLUME AVENUE, HOU	•					Vork Order/Contract/Invoice mber 23690
6. Item:	7. Description	8. Part Number:	9. Eligi	bility*	10. Quantity	11. Serial/ Number:	Batch	12. Status/Work:
3	AIR DATA COMPUTOR;	622-5465-214	622-5465-214 VARIOUS 1 1080				TESTED	
14. Certi	ERTIFIED TO COMPL fies the items identified abov approved design data and are don-approved design data spe	ve were manufactured in cont in a condition for safe opera	formity to:	19. Certifies 12 and d of Feder	14 CFR 43.9 Ret that unless othe escribed in Bloc al Regulations p	turn to Service rwise specified i k 13 was accom	n Block 13 blished in a	egulation specified in Block , the work identified in Bloc secordance with Title 14, Co work, the items are approve
15. Autho	rized Signature:	16. Approval/Authorization No	o.:		n to service.		21. Appro	val/Certificate No.: HE2R869K
17. Name	(Typed or Printed):	18. Date(m/d/y):		22. Name (Typed or Printed	i):	23. Date(r	n/d/y): 8-05
		User/li	nstaller F	Lesponsit	ilities	<u> </u>		
It is import Where the specified	rtant to understand that the existe e user/installer performs work in :	nce of this document alone does	not automat	ically constit	ute authority to i	install the part/co	mponent/a	ssembly.

*Installer must cross-check elegibility with applicable technical data.

2. AUTHORIZED RELEASE CERTIFICATE

23690 - 3

NSN: 0052-00-012-9005

Approving National Aviation

FAA Form 8130-3(6-01)

Fly with

Phone: 713-991

18Y	
JBR61	16
No. UB	06#
R.S. N	K
A. C.F	6

Customer

	STAR FLITE	PS-823B/T	
9	Nomenclature	Part No. 521-1075-0C	Serial No.
#9016	Work Accomplished Overhauled	Bench Checked	Repaired
OKW	Other (Explain) —	,	
L		MAINTENANCE RELEASE	

SERVICEABLE PART

Make 17-7

UNITED BATTERIES AND ACCESSORIES, INC.

CRS #U9BR618Y 7762 BRANIFF HOUSTON, TX 77061

TEL: 713 991-9111 FAX: 713 991-9117 E-MAIL: unitedbatteries@aol.com

FORM #9007

NiCad Battery Pack Work Order # 8323

Customer Starflite Date Received 1/17/05 Customer PO# N838G
Battery Pack/Cell/
Power Supply Mfg. Jet Mfg. Type PS-823B/T Mfg. Part No. 501-1075-06
Serial No. 1310
Pre-Service Hidden Damage Inspection
General Condition PEK HANUFACTURERS SPEC
Receiving Voltage: Terminals 25,80 Battery Pack#1 25,40 ½ Pack
Rattery Pack #2 2.5. 8.8
Outer Case MC Circuit Roard MYS Cells/Cell Cases MYS Hot Spots NEWE
Receptacle Assy. PHS 5 Amp Fuse 10 Amp Fuse
In-Service Hidden Damage Inspection
Capacity Tests, Step 1 through 3
1rst ,13,13,13,13 - PER CELL AVG.
End of Discharge Voltage 22.56.
2nd
End of Discharge Voltage
3rd
End of Discharge Voltage
FINAL CHARGE
4th ,57,57,57,57,57 PER CEW AVG
End of Charge Terminal Voltage 31.4 v
Functional Test PHS PER CHM
Replaced
Remarks
WARRANTY: Months from date of return to service if serviced a minimum of every 6 months.
Work Accomplished By Mulen Kest per CMM TP-202
Work Accomplished By Mules Septe per CMM TP - 202
Inspected By Completion Date /-2/-0]
Inspector Recommendations
The aircraft component, appliance or accessory was inspected in accordance with the current requirements of the
Federal Aviation Administration and is approved for return to service.
Pertinent details are on file at this repair station.
Authorized Inspection Signature Date of Final Inspection for Return to Service Maintenance Release Issued YES or NO
Date of Final Inspection for Return to ServiceMaintenance Release Issued YES or NO

INC. ACCESSORIES. UNITED

U9BR618Y

9

SERVICEABLE PART Make HARA THEW BATT PACK Customer STARFLITE Serial No. Nomenclature Part No. 20-5113 NA. Work Accomplished Bench Checked Overhauled Other (Explain) MAINTENANCE RELEASE

AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON EILE AT THIS REPAIR STATION

Repaired

UNITED BATTERIES AND ACCESSORIES, INC.

CRS #U9BR618Y 7762 BRANIFF HOUSTON, TX 77061

TEL: 713 991-9111 FAX: 713 991-9117

E-MAIL: unitedbatteries@aol.com

FORM #9007

NiCad Battery Pack Work Order # 8324

Customer Starflite Date Received 1/17/05 Customer PO# N83SG Battery Pack/Cell/
Power Supply Mfg. Marathon Mfg. Type 20S113 Mfg. Part No. 20S113
Serial No. N/A
Pre-Service Hidden Damage Inspection
General Condition PER NANUFACTURERS SPEC
Receiving Voltage: Terminals 26.26 Battery Pack#1 26.26 ½ Pack
Dottom Doak #7
Outer Case PHS Circuit Board NA Cells/Cell Cases PHS Hot Spots NONE
Receptacle Assy. PHS 5 Amp Fuse NA 10 Amp Fuse NA
In-Service Hidden Damage Inspection
Capacity Tests, Step 1 through 3 1rst // // // // PER CELL AUG.
End of Discharge Voltage 22,19 2nd
End of Discharge Voltage
3rd 3rd
End of Discharge Voltage
FINAL CHARGE
4th, 58,58,58,58 - PER CELL AUG.
End of Charge Terminal Voltage 31.7
Functional Test
Replaced
Remarks
WARRANTY: Months from date of return to service if serviced a minimum of every 6 months.
Work Accomplished By Mary Sele per CMM AS -95138-9 0388
Inspected By Completion Date Completion Date
Inspector Recommendations
The aircraft component, appliance or accessory was inspected in accordance with the current requirements of the
Federal Aviation Administration and is approved for return to service.
Pertinent details are on file at this repair station
Authorized Inspection Signature Date of Final Inspection for Return to Service Maintenance Release Issued YES of NO
Date of Linds inspection for rectan to parties

INC. CESSORIES, Fly with the AND

713-991-9117 Houston Phone: 713-991-91

U9BR618Y #9016

SERVICEABLE PART He Horizontal Make Ho					
1	Make				
te	4076				
Part No. 1558	Serial No. 38056				
Rench Checked	Renaired				

MAINTENANCE RELEASE

THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON FILE AT THIS REPAIR STATION.

UNDER WORK ORDER NO.

Customer

Nomenclature

Work Accomplished

8326

10

Other (Explain)

FORM #9001-A

NiCad Battery Wet Cell Work Order# 8326

	ttery / Cell Mfg Saft Mfg Part No. 15580 Serial No. 38056										
ttery Type 4076 Volts 24 Amp Hour Rate 36 @# Hours 1 pe of Service: Overhaul I. R. A. N. Charge Other											
					N. A. IV.	Cna	ıge '	Other		· · · · · · · · · · · · · · · · · · ·	——————————————————————————————————————
e-	e-Service Hidden Damage Inspection neral Condition: POR MIFA SPEC										
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ne	ntaala As	a Dai	Cover F	w Cose F	2015	Finas F	2nS	ell Cases	PMS	Hot Spo	ts NOWE
ce	Look A in	sy. 1//	Jane Torn	oinal Volt	2/2 80	Torai	e cettings	unner S	7 100	er 69	hardware
50 0	Leak/Von	dda seco	nage Insp	imiai von artina	2 Den 40 5	Lorqu	ic settings	upper	10 VV	·	
-13	n Droba T	oct a	Civr	vernon –				المانعامية بي			
	(# Cells i				Canacity 5	Tests Ster	os i throu	oh 3	***************************************		
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	_			1)0	-	Com	Lad.				
	12	13	14	15	16	17 6	18	19	20	21	22
							ļ				
			Valtara			mi	nutes Dis	charge R:	ite.	<u></u>	
.1	(# Cells			ALTONOMIC WILLIAM STATE	www.	######################################	nuics Dis	charge re			
a	(# Cens	m series)	1	4	5	6	7	8	9	10	11
	1,20	120	1.20	1.20	1.20	1.20	7 /2c) 18	120	1.20	1.20	
İ	12	18)	14	y 15	16	17	18	19	(20)	21	22
	Volts	Rice C	tell It	17	ĺ .		<u> </u>				
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t	(# Cells	in series)		•	•				*	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	
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	12	13 /2	14	15	16	17	18	19	(20)	21	22
	11011	C. 1	1 1 -1	y y y	7 >)	1	1		i .	
d	of Charge	Termina	l Voltage	32.46 Di	stilled H2	O added	5 cc (harge Ra	te <i>1819</i> -7	05 66	24-5
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71	K / toom			~)			•				\frown
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13	ector Reco	ommenda	tions			-			and the state of t		
e	aircraft co	omponent.	appliance	e or acces	sory was	inspected	in accorda	ince with	current re	quirement	s of the
10	eral Aviati	on Admir	nistration	and is app	roved for	return to	service.				
rti	nent detai	ils are on	file at this	repair sta	ation		7				
(4)				1	/	a //	10				
iŧ	horized b	nspection	Signatur	e///		Tou	/_				
te	te of Final Inspection for Return to Service Maintenance Release Issued YES or NO										

INC. Fly with the Best LNO

77061 Braniff

	4016
15580	Serial No. D81985
Bench Checked	Repaired
n) ————————————————————————————————————	
	1 5 58 0 ■ Bench Checked

SERVICEABLE PART

Make

MAINTENANCE RELEASE THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS INSPECTED IN ACCORDANCE WITH CURRENT

REGULATIONS OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON FILE AT THIS REPAIR STATION.

UNDER WORK ORDER NO

Customer

· FORM #9001-A

NiCad Battery Wet Cell Work Order# 8325

	stomer Starflite Date Received 1/17/05 Customer PO# N83SG ttery / Cell Mfg Saft Mfg Part No. 015580000 Serial No. 081985										
tt	ery / Cell	Mig <u>Sa</u>	H Nig Pa	17 NO. 1	Data	_Seriai N 36 /2# H	0. <u>00170</u>	2			
ttery Type 4076 Volts 24 Amp Hour Rate 36 @# Hours 1 pe of Service: Overhaal I. R. A. N. Charge Other											
			the address of the same of the same of the same of		##+ _F 1 4 +	V 344	· Et	CP 1.EE & F			**************************************
n-	e-Service Hidden Damage Inspection neral Condition PER INFR SPEC ter Case PMS Cover VMS Scal PMS Hardware PMS Terminal Links PMS										
te	r Case T	2/115	Cover L	7115 Se	Din	S Hard	ware P	MS Te	rminal Li	nks PM	-5
CE	ntacle As	sv Pn	S Fille	er Caps \hat{F})/1 S (rings P	ns c	ell Cases	Pins	_Hot Spo	ts Noce
se	Leak/W/	Recei	iving Tern	ninal Volt	s 26:7'	7 Torqu	e settings	upper	7 low	rer <u>62</u>	hardware
·S	ervice Hi	ddeu->	nage Insp	ection	.np.dl.D.commil@Measure.bish.ed.com	necessary t	Ü				\frown
	Service Hidden nage Inspection mp Probe Test 1 Com-										
it	(# Cells	in Series)		(Capacity 1	Tests, Step	os 1 throu	gh 3		·	
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d	(# Cells	in Series)			and the same of th		T 7 .	· ·	£).	10	11
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e	aircraft co	omponent	, appliance	e or acces	sory was i	inspected	in accorda	ance with	current re	quirement	ts of the
d	eral Aviati	ion Admir	nistration	and is app	roved for	return to	service.				
rt	inent deta	ils are on	file at this	repair stà	ition	7					
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	ithorized Inspection Signature Maintenance Release Issued YES or NO										